## SIMULATION OF NYLON-6 POLYMERIZATION IN TUBULAR REACTORS

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DEPARTMENT OF CHEMICAL ENGINEERING
INDIAN INSTITUTE OF TECHNOLOGY KANPUR
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# SIMULATION OF NYLON-6 POLYMERIZATION IN TUBULAR REACTORS

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MASTER OF TECHNOLOGY

DEBASIS PAL

to the

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INDIAN INSTITUTE OF TECHNOLOGY KANPUR
JANUARY, 1988

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#### CERTIFICATE

This is to certify that the present work 'SIMULATION OF NYLON-6 POLYMERIZATION IN TUBULAR REACTORS' has been carried out by Mr. Debasis Pal under my supervision and that this has not been submitted elsewhere for a degree.

January 1988

( Santosh K. Gupta )

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- Debasis Pal

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#### NOMENCLATURE

$A_{i}^{C}$ , $A_{i}^{C}$	Frequency factors
A <sub>i,j</sub>	Matrix for the first derivative in orthogonal collocation
B <sub>i,j</sub>	Matrix for the second derivative in orthogonal collocation
c <sub>1</sub>	€-caprolactam
c <sub>2</sub>	Cyclic dimer
$c_p$	Specific heat, Kcal/Kg-K
(c <sub>p</sub> ) <sub>j</sub>	Specific heat at the jth grid point, Kcal/Kg-K
$E_{i}^{C}$ , $E_{i}^{C}$	Activation energy
h	Overall heat transfer coefficient, Kcal/m <sup>2</sup> -K-hr
Δ H <sub>i</sub>	Enthalpy of reaction, cal/mol
k	Thermal conductivity, Kcal/m-K-hr
k <sub>i</sub>	Forward rate constant
k <sub>i</sub>	Backward rate constant
K	Equilibrium constant
N	Number of internal grid points (excluding the point at the wall)
<b>P</b> <sub>1</sub>	Aminocaproic acid
$P_{\mathbf{n}}$	Polymeric species of chain length n
P <sub>n</sub>	Symmetric Legendre polynomial of degree n
ΔQ	Overall heat of generation, Kcal/Kg-hr
r	Radius, m
rj	Radial distance of jth grid point, m
Δr	Distance between two grid points in finite difference (equal spacing), m
R	Reactor radius, m

R Universal gas constant, cal/mol-K

 $\Delta S_{i}$  Entropy of reaction, eu

t Average residence time, hr

T Temperature, K

Average temperature at any cross-section, K

T<sub>J</sub> Coolant temperature in jacket, K

U; ith variable at jth grid point

v Axial velocity, m/hr

v av Average axial velocity, m/hr

W Water

Weightage vector in orthogonal collocation

x Non-dimensional radius

 $\mathbf{x}_{i}$  Non-dimensional radius at jth grid point

z Axial distance, m

#### Greek Letters:

 $\lambda_{k}$  kth moment for the polymeric species, mol/Kg

p Density, Kg/m<sup>3</sup>

 $\rho_i$  Density at jth grid point, Kg/m<sup>3</sup>

μ<sub>n</sub> Number average chain length

 $\vec{u}_n$  Average value of the number average chain length at any cross section

#### Subscript:

O Initial value in the feed

#### Symbol:

[ ] Concentration of a species, mol/Kg

#### ABSTRACT

This work presents a comprehensive computer simulation for the hydrolytic polymerization of € - caprolactam without water removal in a continuous tubular reactor. This incorporates the existence of a laminar (Hagen-Poiseuille) velocity profile and radial thermal diffusion. Solution of the balance equations are carried out using two types of finite difference techniques as well as the method of orthogonal collocation coupled with GEAR's method for ODE's. The simulation results are obtained for isothermal, adiabatic and non-adiabatic operations for both plug and laminar flow velocity profiles. Some of the results were compared with available results for simpler cases. influence of various operational variables on the reactor performance was studied. A comparison is made of the different The first method methods used and their limitations discussed. of finite difference (method I) was found to be superior to the second one, though both of them gave the same results for higher number of grid points. The orthogonal collocation technique gave good results except for the laminar flow adiabatic and non-adiabatic cases where the temperature gradients near the wall were steep. An improvement is suggested by including the finite element technique.

#### CHAPTER 1

#### INTRODUCTION

The hydrolytic polymerization of \$\varcetarrow{\varcetarr

The work on nylon-6 polymerization falls broadly into two categories. The emphasis in the early work was primarily on the determination of the polymerization mechanism and of the rate and equilibrium constants for the various reactions. These have been reviewed by Reimschuessel<sup>1</sup>. More recently the emphasis has shifted to the study of physical processes like heat transfer, mass transfer etc., associated with chemical reaction in large scale reactors, and to the actual modeling and optimization of industrial reactors. Tai and Tagawa<sup>2</sup>, and Kumar and Gupta<sup>3,4</sup> have reviewed these aspects.

The reaction mechanism comprises of three major reactions, namely, the ring opening of caprolactam by water to form aminocaproic acid, and chain and step growth reactions. In addition, there are several important side reactions. Among these is the formation of cyclic oligomers, desamination and peroxidation of caprolactam<sup>1</sup>. The most important side reactions are those associated with cyclic oligomers, since their presence in the

product causes problems in its processing (e.g., in spinning and molding). The kinetic scheme considered in this work is shown in Table I. This includes the main reactions and the reactions associated with the cyclic dimer. The other cyclization reactions are omitted since their rate constants are not yet available. Moreover, it is well known<sup>3,5</sup> that the formation of the cyclic dimer predominates, and can be used as a first-order approximation of the total cyclics present.

The reactions are known to be catalyzed by the carboxyl end groups present in the reaction mass. The apparent rate constants are of the form

$$k_i = k_i^0 + k_i^C \quad [-COOH]$$
 (1)

with Arrhenius forms being used for  $k_i^O$  and  $k_i^C$ . All the reactions in Table I are reversible in nature and the temperature dependency of the equilibrium constants are given by standard thermodynamic relations. The rate and equilibrium constants for the reactions of Table I are given in Table II. These are based on results from a series of experiments carried out by Tai et al., using a nonlinear regression analysis.

Various kinds of reactors, e.g., batch, tubular, continuous-flow-stirred-tank reactors (CSTRs) and combinations thereof, are employed in industry to manufacture nylon-6.

#### Table I

#### Kinetic Scheme For Nylon-6 Polymerization

1. Ring Opening:

$$c_1 + w = \frac{k_1}{k_1' = k_1/K_1} p_1$$

2. Polycondensation:

$$P_n + P_m = \frac{k_2}{k_2' = k_2/K_2}$$
  $P_{n+m} + W ; n, m = 1,2,3 ...$ 

3. Polyaddition:

$$P_n + C_1 \xrightarrow{k_2} P_{n+1} ; n=1,2,...$$

4. Ring Opening of Cyclic Dimer:

$$c_2 + w \xrightarrow{k_4} p_2$$

$$k_4' = k_4/K_4$$

5. Polyaddition of Cyclic Dimer:

$$P_n + C_2 = \frac{k_5}{k_5^2 = k_5/K_5}$$
  $P_{n+2}$ ;  $n=1,2,...$ 

Contd....

#### Table I (Continued)

$$C_1: H-N-(CH_2)_5-C$$
 (\varepsilon - caprolactam)

$$C_2: H-N-(CH_2)_5 - C-N-(CH_2)_5 - C(cyclic dimer)$$

$$P_n: H \stackrel{H}{=} N - (CH_2)_5 - \stackrel{O}{=} \frac{1}{3} OH$$
 (Polymer: Nylon-6)

# Table II

Rate and Equilibrium Constants for Nylon-6 Polymerization

a<sub>Ref 2</sub>

Computer simulations based on the mathematical modeling of the polymerization processes in these reactors or their combinations offer information which is of paramount importance for quality control, process control and operational optimization of existing plants as well as in the design of new plants. Several studies have been reported on the simulation of both ideal reactors<sup>7-9</sup> and some common industrial reactors 10-13.

The present work focusses on the simulation of continuous tubular reactors with radial variations of temperature and concentrations accounted for. In most of the studies on tubular reactors reported till now, radial variations of temperature (and hence, of the concentrations) have not been considered. This was probably to keep the analysis simple. The only study wherein such radial profiles are indeed computed is the one by Tai et al. 7. These workers have used the two dimensional (in the radial and axial directions) finite difference computational technique to solve their balance equations. However, they have also presented results on several other types of reactors and reactor combinations, and so their work on tubular reactors is slightly less comprehensive than what one would like to have. Moreover, our initial attempts on the optimization of such reactors has revealed the need for faster and more efficient computational techniques to integrate the modeling equations for such reactors, in order to cut down on the

total computational costs. This work was undertaken to satisfy these needs, and presents simulation results using two types of finite difference algorithms, as well as the orthogonal collocation technique. Results are generated for several conditions, e.g., for both plug and laminar flow velocity profiles in the tubular reactor and for both adiabatic and non-adiabatic operations. A detailed comparison is made of the results obtained with various computational techniques and model assumptions.

#### CHAPTER 2

#### FORMULATION

The reaction mechanism considered here is given in Table I with the rate and equilibrium constants as shown in Table II. The rates of formation of various species in the reaction mixture and those of the first two moments,  $\lambda_0$  and  $\lambda_1$ , are listed in Table III along with the rate of heat generation. In general, the  $k^{th}$  moment is defined as:

$$\lambda_{k} = \sum_{n=1}^{\infty} n^{k} [P_{n}] ; k = 0,1,2....$$
 (2)

The widely accepted use of moments instead of individual concentrations of polymer molecules of various chain lengths helps reduce the number of equations ( for the individual species) to be solved simultaneously, and gives almost the same results  $^{3,17-19}$ . At the same time, information about the first two or three moments are usually sufficient to know the nature of the molecular weight distribution in the product polymer. For example, the number average chain length ( $\mu_n$ ) is obtained from the first two moments,

$$\mu_{\mathbf{n}} = \lambda_{1} / \lambda_{0} \tag{3}$$

Table IV shows the mass and heat balance equations as applied to a tubular reactor (see Fig.1) having a laminar (Hagen-Poiseuille) flow profile given by

#### Table III

#### Species Rate of Formation and Heat Generation Rate Terms

$$R_{C_{1}} = -k_{1} [C_{1}] [W] + k_{1}' [P_{1}] - k_{3} [C_{1}] \lambda_{0} + k_{3}' (\lambda_{0} - [P_{1}])$$
 (a)

$$R_{P_{1}} = k_{1} [C_{1}][W] - k_{1}'[P_{1}] - 2k_{2} [P_{1}] \lambda_{0} + 2k_{2}'[W] (\lambda_{0} - [P_{1}])$$

$$-k_{3} [P_{1}][C_{1}] + k_{3}' [P_{2}] - k_{5} [P_{1}][C_{2}] + k_{5}' [P_{3}]$$
(b)

$$R_{\lambda_{0}} = k_{1} [C_{1}][W] - k_{1}'[P_{1}] - k_{2} \lambda_{0}^{2} + k_{2}'[W] (\lambda_{1} - \lambda_{0})$$

$$+ k_{4} [W][C_{2}] - k_{4}' [P_{2}]$$
(c)

$$R_{\lambda_{1}} = k_{1} [C_{1}][W] - k_{1}' [P_{1}] + k_{3} [C_{2}] \lambda_{0} - k_{3}' (\lambda_{0} - [P_{1}])$$

$$+ 2k_{5} [C_{2}] \lambda_{0} - 2k_{5}' (\lambda_{0} - [P_{1}] - [P_{2}])$$

$$+ 2k_{4} [W][C_{2}] - 2k_{4}' [P_{2}]$$
(d)

$$R_{C_2} = -k_4 [C_2][W] + k_4 [P_2] - k_5 [C_2] \lambda_0 + k_5 (\lambda_0 - [P_1] - [P_2])$$
(e)

$$R_{W} = -k_{1} [C_{1}][W] + k_{1}[P_{1}] + k_{2} \lambda_{0}^{2} - k_{2}[W](\lambda_{1} - \lambda_{0})$$

$$-k_{4} [C_{2}][W] + k_{4}[P_{2}]$$
(f)

△ Q = Heat generation rate (Kcal/kg-hr)

$$= \sum_{i=1}^{5} -(\Delta H)_{i} R_{i}/10^{3}$$
 (g)

#### Table III (Continued)

where.

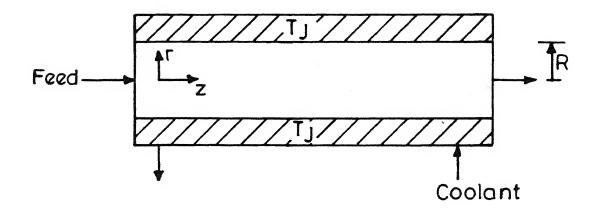
$$R_1 = k_1 [C_1][W] - k_1' [P_1]$$
 (h)

$$R_2 = k_2 \lambda_0^2 - k_2' [W] (\lambda_1 - \lambda_0)$$
 (i)

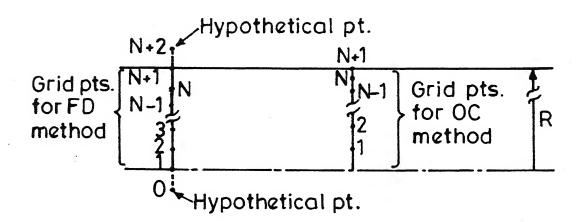
$$R_3 = k_3 [C_1] \lambda_0 - k_3' (\lambda_0 - [P_1])$$
 (j)

$$R_4 = k_4 [C_2][W] - k_4 [P_2]$$
 (k)

$$R_5 = k_5 [C_2] \lambda_0 - k_5' (\lambda_0 - [P_1] - [P_2])$$
 (1)



(a)



(b)

Fig.1

- (a) Schematic presentation of the tubular reactor with cooling jacket
- (b) Grid points for finite difference (methods I & II) and orthogonal collocation methods (method IV)

$$v_z = 2v_{av} (1-r^2/R^2)$$
 (4)

The average residence time, t, is given by

$$t = z/v_{av}$$
 (5)

where z is the distance along the reactor. In writing these equations, the diffusion of all the components has been neglected and only thermal diffusion in the radial direction has been considered. While the diffusion of heavy polymer molecules may indeed be negligible, it may not be quite justified to neglect the diffusion of the lower molecular weight components, particularly, the monomer and water. This assumption is made for two reasons firstly because precise data on the diffusion coefficients are still not available 14, and secondly as the radial diffusion in the absence of vaporization is expected to be a second order effect, and could be neglected at least for optimization studies. Tai et al. have also made similar approximations. It may be pointed out here that recently some work has been reported by Hamer and Ray 20 on the detailed modeling of continuous tubular polymerization reactors incorporating radial diffusion of mass as well as property and velocity profile changes. Work along these lines for nylon-6 reactors could be carried out, but since our long range focus was on optimization studies, we did not pursue this approach.

The following correlations<sup>7,21</sup> have been used for the various thermophysical properties

thermal

density : 
$$\int (Kg/m^3) = 1000 \{ 1.0065 + 0.0123 [c_1] + (T(K) - 495) (0.00035 + 0.00007 [c_1]) \}^{-1}$$
 (6b)

specific heat : 
$$C_p \left( \frac{Kcal}{Kg-K} \right) = 0.6593 \left[ \frac{C_1}{C_1} \right]_o + \left\{ 1 - \left[ \frac{C_1}{C_1} \right]_o \right\} \times \left\{ 0.4861 + 0.000337 T(K) \right\}$$
 (6c)

The boundary condition at the wall of the reactor for the heat balance equation (eqn.h. Table IV) is obtained by considering a coolant at a constant temperature,  $T_J({}^{\circ}K)$ , circulated in a jacket enclosing the reactor (Fig. 1a), the overall heat transfer coefficient being h. The symmetry condition at r=0 is given by eqn.i, Table IV.

A radial temperature gradient will be established in the reactor at any cross-section along the length because of the presence of the velocity profile as well as the heat flux through the wall. As a consequence, the concentrations of all the species will also have a radial variation. Hence, the variables  $[C_1]$ ,  $[P_1]$ ,  $\lambda_0$ ,  $\lambda_1$ ,  $[C_2]$ , [W], and T are all functions of r and t in general, and can be represented as  $[C_1]$  (r,t), etc. To obtain all these profiles, the set of differential equations (eqns. a-g, Table IV) need to be

#### Table IV

### Mass and Heat Balance Equations in a Tubular Reactor with Laminar Flow Profile

Mass Balance:

$$2(1 - r^2/R^2) \frac{d[c_1]}{dt} = R_{c_1}$$
 (a)

$$2(1 - r^2/R^2)$$
  $\frac{d[P_1]}{dt} = R_{P_1}$  (b)

$$2(1 - r^2/R^2) \frac{d\lambda_0}{dt} = R\lambda_0$$
 (c)

$$2(1 - r^2/R^2) \qquad \frac{d\lambda_1}{dt} = R\lambda_1 \tag{d}$$

$$2(1 - r^2/R^2)$$
  $\frac{d[C_2]}{dt} = R_{C_2}$  (e)

$$2(1 - r^2/R^2) \frac{d[w]}{dt} = R_W$$
 (f)

Heat Balance:

$$2(1 - r^2/R^2) f c_p \frac{\partial T}{\partial t} = \frac{k}{r} \frac{\partial}{\partial r} \left[ r \frac{\partial T}{\partial r} \right] + f(\Delta Q)$$

(g)

$$-k \left| \frac{\partial T}{\partial r} \right|_{r=R} = h (T - T_J)$$
 (h)

$$\frac{\partial T}{\partial r} \bigg|_{r=0} = 0 \tag{i}$$

$$t = z/v_{av}$$

integrated simultaneously with the boundary and symmetry conditions (eqns.h-i, Table IV).

A close examination of the equations of Table IV alongwith the rate expressions of Table III reveals that the equations for  $[C_1]$ ,  $[P_1]$ ,  $\lambda_0$ ,  $\lambda_1$ ,  $[C_2]$ , [W] and T do not involve any other variable other than  $[P_2]$  and  $[P_3]$ . The equations for  $[P_2]$  and  $[P_3]$  will similarly involve  $[P_4]$  and  $[P_5]$ . To break this hierarchy of equations and to make them solvable, a simplifying assumption, called the closure approximation, is employed. This is as follows:

$$[P_3] = [P_2] = [P_1]$$
 (7)

This was first used by Reimschuessel<sup>1</sup>. Tai et al.<sup>22</sup> have found that the final results are insensitive to this assumption. The equations of Table IV with eqn.7 now comprise a complete set of coupled differential equations which need to be integrated numerically.

Two techniques are employed in this work to convert the partial differential equation - PDE (eqn.g, Table IV) into an ordinary differential equation - ODE. These are the finite difference (FD) and the orthogonal collocation (OC) methods. The basic principles of these methods are explained in Ref. 23.

To use the FD method, the radial distance,  $0 \leqslant r \leqslant R$  is first divided into N equal parts, of length  $\triangle r$  each,

using (N+1) grid points numbered from 1 to N+1 (Fig. 1b). Thus the jth radial grid point at  $r=r_j$  is at  $r=(j-1)(\triangle r)$ . The variables at  $r_j$  are renamed for the sake of convenience as follows:

$$U_{1,j}(t) \equiv [C_{1}](r_{j},t); \quad U_{2,j}(t) \equiv [P_{1}](r_{j},t);$$

$$U_{3,j}(t) \equiv \lambda_{0}(r_{j},t); \quad U_{4,j}(t) \equiv \lambda_{1}(r_{j},t);$$

$$U_{5,j}(t) \equiv [C_{2}](r_{j},t); \quad U_{6,j}(t) \equiv [W](r_{j},t);$$

$$U_{7,j}(t) \equiv T(r_{j},t); \quad j = 1,2,...,N+1.$$

$$(8)$$

The radial derivatives in eqn.g, Table IV, the boundary condition (eqn.h, Table IV) and the symmetry condition (eqn.i, Table IV) are converted into algebraic expressions using the FD technique 23. This has been done in two different ways. In the first method (method:I), the Laplacian in the heat balance (eqn.g, Table IV) is broken up into two parts to give

$$2(1-r^2/R^2)fc_p\frac{\partial T}{\partial t} = \frac{k}{r}\frac{\partial T}{\partial r} + k\frac{\partial^2 T}{\partial r^2} + f(\Delta Q)$$
 (9)

The finite difference formulae<sup>23</sup> for the first and second derivatives (mid-point formulae) are then applied to give following ODE corresponding to the jth point:

$$2(1-r_{j}^{2}/R^{2}) \int_{j}^{r} (C_{p})_{j} \frac{dU_{7,j}}{dt}$$

$$= k \left[ \frac{1}{r_{j}} \frac{U_{7,j+1} - U_{7,j-1}}{2(\Delta r)} + \frac{U_{7,j+1} - 2U_{7,j} + U_{7,j-1}}{(\Delta r)^{2}} \right]$$

$$+ \int_{j}^{r} (\Delta \Omega)_{j}; \quad j=2,...,N$$
(10)

At the center, i.e., at j=1 (r=0) eqn.10 presents a problem, because it involves  $U_7$  at an undefined point, j=0, as well as involves division by zero ( $r_1$ =0). This is resolved by using L'Hopital's rule as well as the symmetry condition in its FD form  $^{23}$ :

$$\frac{U_{7,2} - U_{7,0}}{2(\Delta r)} = 0; \text{ or, } U_{7,0} = U_{7,2}$$
 (11)

Here, U<sub>7.0</sub> is the temperature at a hypothetical point, j=0 (Fig. 1b). The resulting equation corresponding to j=1 is,

$$2(1-r_1^2/R^2) f_1(c_p)_1 \frac{dU_{7,1}}{dt}$$

$$= \frac{4k}{(\Delta r)^2} (U_{7,2} - U_{7,1}) + f_1(\Delta Q)_1$$
 (12)

Similarly eqn. 10 for j=N+1 (r=R, at the wall) involves  $U_{7,N+2}$ , the temperature at another hypothetical point. This is resolved by using the FD form of the boundary condition (eqn.h, Table IV):

$$U_{7,N+2} = U_{7,N} - \frac{2(\Delta r)h}{k} (U_{7,N+1} - T_{J})$$
 (13)

This is used in eqn.10 for j=N+1. The set of mass balance equations can be transformed to the FD form quite easily. Thus, 7(N+1) ordinary differential equations have been generated, which are given in Table V. The number of variables to be solved for are also 7(N+1). These ODEs are solved using a GEAR's package 23, which has a built-in step control algorithm and is particularly useful for stiff systems. The NAG Library Routine DO2EBF was used for this purpose in a DEC 1090 system.

The second approach of the finite difference technique (method II) is very similar to method I except at the boundary points, j=1 and N+1. To apply the FD technique to the boundary and symmetry condition, one-sided first derivative formulae (ref. 23, p.68), which have the same degree of accuracy as the mid-point formulae used in method I, have been used. The symmetry condition at r=0 is transformed into:

$$U_{7,1} = \frac{4}{3} U_{7,2} - \frac{1}{3} U_{7,3}$$
 (14)

and the boundary condition at r=R gives,

$$U_{7,N+1} = \frac{hT_{J} - \frac{k}{2(\Delta r)} (U_{7,N-1} - 4U_{7,N})}{h + \frac{3k}{2(\Delta r)}}$$
(15)

Use of the FD form of the heat balance equation at the center and boundary points is avoided by using eqn. 14 and 15.

#### Table V

## Set of Differential Equations After Applying Method I Of Finite Difference Technique

$$2(1 - r_j^2/R^2)$$
  $\frac{dU_{1,j}}{dt} = (R_{U_1})_j, j = 1, \dots, N+1$  (a)

$$2(1 - r_j^2/R^2)$$
  $\frac{dU_{2,j}}{dt} = (R_{U_2})_j, j = 1, \dots, N+1$  (b)

$$2(1 - r_j^2/R^2) \frac{dU_6}{dt}, j = (R_{U_6})_j, j = 1, \dots, N+1$$
 (c)

$$2(1 - r_1^2/R^2)f_1(c_p)_1 \frac{du_{7,1}}{dt} = \frac{4k}{(Ar)^2}(u_{7,2}-u_{7,1}) + f_1(AQ)_1$$
 (d)

$$2(1 - r_j^2/R^2) P_j(c_p)_j \frac{dU_{7,j}}{dt}$$

$$= k \left[ \frac{1}{r_{j}} \frac{U_{7,j+1} - U_{7,j-1}}{2(\Delta r)} + \frac{U_{7,j+1} - 2U_{7,j} + U_{7,j-1}}{(\Delta r)^{2}} \right] + \mathcal{P}_{j} (\Delta Q)_{j} \quad j = 2, \dots, N+1$$

with

$$U_{7,N+2} = U_{7,N} - \frac{2(\Delta_r) h}{k} (U_{7,N+1} - T_J)$$
 (e)

dependence of variables on t is not explicitly mentioned.

The resultant set of ODEs consists of eqn. a-c, Table V for j=1,...,N+1 and eqn.e, Table V for j=2,...,N. The total number of ODEs is 7(N+1)-2 and the number of variables is also 7(N+1)-2. The other two variables, i.e., the temperatures at the center and at the external boundary point, can be obtained from eqn.14 and 15. The same GEAR's package is used for integrating the set of equations as well.

A compact computer package for integrating PDEs (NAG Library Routine DO3PBF) was also used (method III). This package consists of a combination of the finite difference technique and the GEAR's package to solve a set of PDEs with appropriate boundary conditions. As discussed later, method I and III are essentially the same. However, method III has an additional flexibility in the sense that it can be used for any type of unequal grid spacing, upon changing an input integer, IMESH. The value used for our case was 4, which results in having more grid points near the external boundary (where the gradients are expected to be steeper) according to formula:

$$r_i = R \sin \left( \frac{\pi(i-1)}{2N} \right) \tag{16}$$

This method was used to generate various simulation results.

Another technique tried was the method of orthogonal collocation. The ability of this techniques to give results

of greater accuracy with less computational effort under certain conditions, is well established  $^{23}$ . However, no work has been reported till now employing this efficient method for the simulation of nylon-6 tubular reactors with radial gradients. In this method (method IV), the variables are approximated as a series of orthogonal Legendre polynomials. These polynomials are symmetric in nature with respect to the independent variable which is x = r/R. The polynomials themselves are obtained from the orthogonality conditions e.g.,

$$\int_{0}^{1} W(x^{2}) P_{k}'(x^{2}) P_{m}'(x^{2}) x^{a-1} dx = 0,$$

$$k \leq m-1$$
(17)

where,  $W(x^2)$  is a weightage function which has been taken as  $(1-x^2)$  in the present case. 'a' is the geometry factor and has the value of 2 for the cylindrical geometry. Once  $W(x^2)$  and 'a' are fixed, all these polynomials can be obtained. The polynomial with the highest degree to be considered if  $P_N(x^2)$  where N is the number of interior collocation points (excluding the external boundary). The roots of  $P_N(x^2)$  are taken as the interior collocation points. Now the matrices  $A_{i,j}$  and  $B_{i,j}$  may be formed  $P_N(x^2)$  from the collocation points so that, for any dependent variable  $P_N(x^2)$ 

$$\frac{dy}{dx} \Big|_{\text{at ith point}} = \sum_{j=1}^{N+1} A_{j,j} y_{j}$$
 (18)

and

$$\frac{d^2y}{dx^2} \Big|_{\text{at ith point}} = \sum_{j=1}^{N+1} B_{i,j} y_j$$
 (19)

Using eqn.18 in the boundary condition (eqn.h, Table IV) for our case, after nondimensionalizing the independent variable r in terms of x, we get,

$$U_{7,N+1} = \frac{-\frac{k}{R} \sum_{j=1}^{N} A_{N+1,j} U_{7,j} + hT_{J}}{h + \frac{k}{R} A_{N+1,N+1}}$$
(20)

Thus, there is no need to apply the heat balance at the boundary. The symmetry condition need not be considered here as, this is automatically taken into account by the use of the symmetric polynomials. The heat balance (eqn.g, Table IV) after being written at the jth collocation point gives,

$$2(1-x_{j}^{2}) \int_{j}^{c} (c_{p})_{j} \frac{dU_{7,j}}{dt}$$

$$= (\frac{k}{R^{2}}) \sum_{i=1}^{N+1} B_{j,i} U_{7,i} + \int_{j}^{c} (\Delta 0)_{j},$$

$$j=1,...N \qquad (21)$$

After substituting eqn. 20 in eqn. 21 the final form of the heat balance is:

$$2(1-x_{j}^{2}) \int_{j}^{c} (C_{p})_{j} \frac{dU_{7,j}}{dt}$$

$$= (\frac{k}{R^{2}}) \sum_{i=1}^{N} B_{j,i} U_{7,i} + \frac{(-\frac{k}{R}) \sum_{i=1}^{N} A_{N+1,i} U_{7,i} + hT_{J}}{(h + \frac{k}{R}) A_{N+1,N+1})} + \int_{j}^{c} (\Delta \Omega)_{j}$$

$$j=1,...,N \qquad (22)$$

The mass balance equations (eqn. a-f. Table IV) can again be applied at all the (N+1) collocation points. The resultant set of [7(N+1)-1] ODEs are listed in Table VI. From these, the [7(N+1)-1] variables (other than  $U_{7,N+1}$ , the temperature at the boundary) can be obtained by the use of GEAR's method with appropriate initial conditions. The temperature at the boundary ( $U_{7,N+1}$ ) is obtained by using eqn. 20.

In all the numerical techniques, the variables obtained at the radial grid points at any tame used to get the average values. The averages can be computed using the 'cup mixing rule' to give, for any concentration and moment,

$$(U_{i})_{av}^{(t)} = \frac{\int_{0}^{R} r(1-r^{2}/R^{2})U_{i} f(r) dr}{\int_{0}^{R} r(1-r^{2}/R^{2}) f(r) dr} ,$$
for i=1,...,6 (23)

#### Table VI

#### Set of ODEs Using Orthogonal Collocation

$$2(1 - x_j^2) - \frac{dU_{i,j}}{dt} = (R_{U_1})_j, j = 1, \dots, N+1$$
 (a)

$$2(1 - x_{j}^{2}) \frac{dU_{2,j}}{dt} = (R_{U_{2}})_{j}, \quad j = 1, \dots, N+1$$
 (b)
$$\vdots \qquad \vdots \qquad \vdots \qquad \vdots$$

$$2(1 - x_{j}^{2}) \frac{dU_{6,j}}{dt} = (R_{U_{6}})_{j}, \quad j = 1, \dots, N+1$$
 (c)

$$2(1-x_j^2)$$
  $\frac{dU_{6,j}}{dt} = (R_{U_6})_j$ ,  $j = 1, \dots, N+1$  (c)

$$2(1 - x_j^2) f_j(c_p)_j \frac{du_{7,j}}{dt} = (\frac{k}{R^2}) \sum_{i=1}^{N} b_{j,i} u_{7,i} +$$

$$\frac{\left(-\frac{k}{R}\sum_{i=1}^{N}A_{N+1,i}U_{7,i}+hT_{J}\right)}{\left(h+\frac{k}{R}A_{N+1,N+1}\right)} + P_{j}(\Delta Q)_{j};$$

$$j = 1, .....$$
 (d)

and for temperature,

$$(U_7)_{av}(t) = \frac{\int_0^R r(1-r^2/R^2)U_7 f(r) c_p(r) dr}{\int_0^R r(1-r^2/R^2) f(r) c_p(r) dr}$$
(24)

In the first three methods the integrals are computed by employing a third order finite difference formula given by Gill and Miller 24. The NAG Library Routine DOIGAF is used for this purpose. In the orthogonal collocation method, the integrals are evaluated using the weightage vector generated from orthogonal polynomials 23. The final form is as follows:

for concentration and moments,

$$(U_{i})_{av} = \frac{\sum_{j=1}^{N+1} W_{j}(1-x_{j}^{2}) U_{i,j} f_{j}}{\sum_{j=1}^{N+1} W_{j}(1-x_{j}^{2}) f_{j}}$$

$$= \frac{\sum_{j=1}^{N+1} W_{j}(1-x_{j}^{2}) f_{j}}{\sum_{j=1}^{N+1} W_{j}(1-x_{j}^{2}) f_{j}}$$

$$= 1, \dots, 6. \quad (25)$$

and, for temperature,

$$(\mathbf{U}_{7})_{av} = \frac{\sum_{j=1}^{N+1} w_{j} (1-x_{j}^{2}) u_{7,j} \beta_{j} (c_{p})_{j}}{\sum_{j=1}^{N+1} w_{j} (1-x_{j}^{2}) \beta_{j} (c_{p})_{j}}$$

$$(26)$$

In laminar flow, the residance time is infinity at the wall. To avoid numerical problems, a small slip  $(v_z(r=R)/v_{av}=10^{-4})$  was assumed at the wall. The numerical results were found to

be relatively unaffected by changing the dimensionless velocity around 10<sup>-4</sup>. The inlet condition (t=0) at the wall (r=R) was also modified. The equilibrium conversion values of isothermal runs at 250°C were fed in as values at r=R, t=0. This reduced the computational time significantly and avoided convergence problems. Any change in these input values was found not to affect the results, but led to longer computational times.

## CHAPTER 3

## RESULTS AND DISCUSSION

A general computer program was made. Results for isothermal conditions could be obtained by putting  $\Delta$  Q=0 in the computer program. Similarly, adiabatic operation could be simulated by putting the heat transfer coefficient, h, as 0. The program could be run using either plug flow ( $v_z$  = Constant= $v_{av}$ ) or parabolic flow (eqn.4) profiles. A typical run for t = 20 hrs and 10 FD points took 3 min 45 sec and for 10 OC points it took 2 min 36 sec on a DEC 1090 system.

The computer programs for all the four methods described previously were checked with results available in the literature. 10,25 The programs for methods I-IV were run for the simple case of an isothermal plug flow reactor with the input conditions as:

and for temperatures of 230°C, 240°C, 250°C, 260°C and 270°C.

The results were found to be in complete agreement with those of Ray and Gupta, thus confirming the correctness of the computer programs (except the heat balance equations). The programs were also run for an adiabatic plug flow reactor with the feed coming from the top of a VK column 10. The results were found to match

those of Gupta and Gandhi<sup>10</sup>. This further confirmed the correctness of our computer programs (including the heat balance equations).

Having developed confidence in the computer packages prepared, detailed temperature and concentration profiles for several conditions were generated. The feed conditions were the same as in eqn.27. Table VII gives the various flow and heat transfer conditions used. Figs.2 to 17 present profiles for monomer conversion, number average chain length( $\mu_n$ ), temperature and cyclic dimer concentration at various values of t ( $\equiv$  z/v<sub>av</sub>). For heat transfer coefficient, h, a typical value 11 of 5.0 kcal/m²-K-hr was used. The existence of distinct concentration and temperature profiles are quite evident from these figures.

Figs. 2 to 5 show the profiles at t = 5hr for various flow and heat transfer conditions (Table VII). The laminar flow case (curves 4,5 & 6) is the most interesting. Since the velocity near the center is higher, the residence time of a fluid element near the center (at the same t) is smaller when compared to that of a fluid element near the wall. Thus the reaction is close to equilibrium at the wall. A maximum in the temperature slightly away from the wall is observed (curves 5 & 6, Fig-2) at t=5hr. This represents a combination of higher heat geneartion (due to higher near equilibrium conversion) near the wall as wêll as heat

Table VII

Various Flow And Heat Transfer Conditions Used In Simulation Runs

Run No•	Type of Flow	Heat Transfer Condition	Heat Transfer Feed Coefficient, Temps, h(Kcal/m²-K-hr) T <sub>O</sub> (C)	Feed Temps, To (C)	Coolant Temp., T <sub>J</sub> (°C)
	Plug	Isothermal	l	250	ŧ
2	Plug	Adiabatic	0	250	i ,
en '	Plug	Non-Adiabatic	5.0	250	760
4	Laminar	Laminar Isothermal	ſ	250	ı
2	Laminar	Laminar Adiabatic	0	250	i
9	Laminar	Laminar Non-adiabatic	5.0	250	260

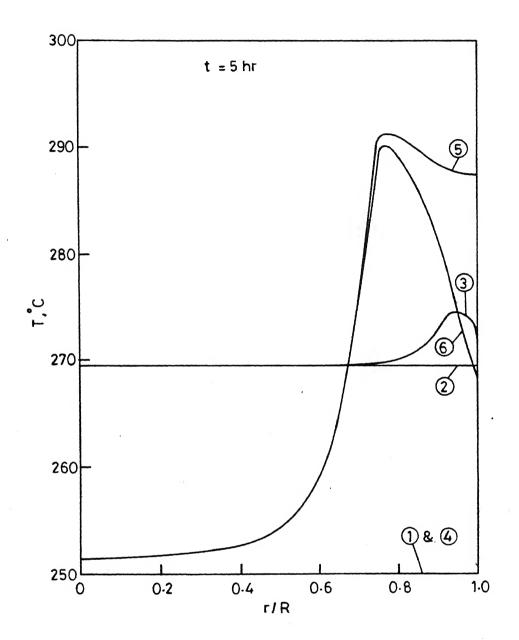


Fig. 2.

Radial profile for temperature at t=5 hr for various flow and heat transfer conditions (given in Table VII). Feed condition is as per eqn. 27.

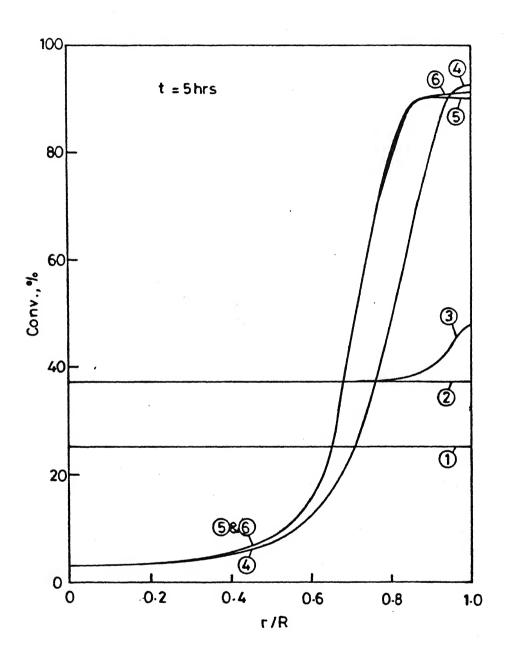


Fig. 3.

Radial profile for conversion at t=5 hr for various flow and heat transfer conditions (given in Table VII). Feed condition is as per eqn. 27.

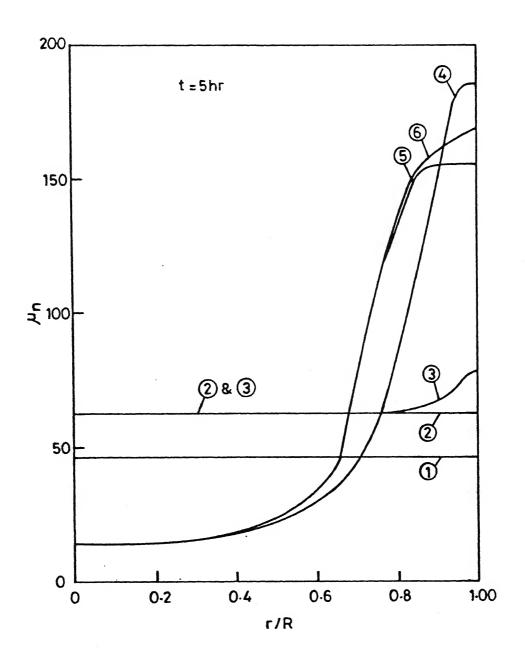


Fig. 4. Radial profile for  $\bigwedge_n$  at t=5 hr for various flow and heat transfer conditions (given in Table VII). Feed is as per eqn. 27.

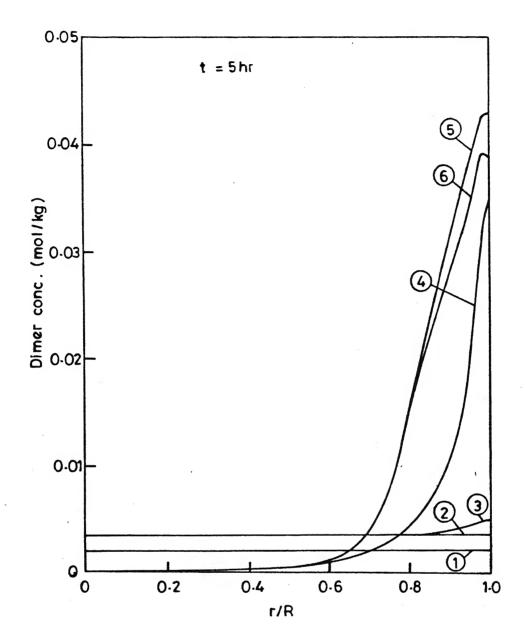


Fig.5. Radial profile for  $[C_2]$  at t=5 hr for various flow and heat transfer conditions (given in Table VII). Feed condition is as per eqn. 27.

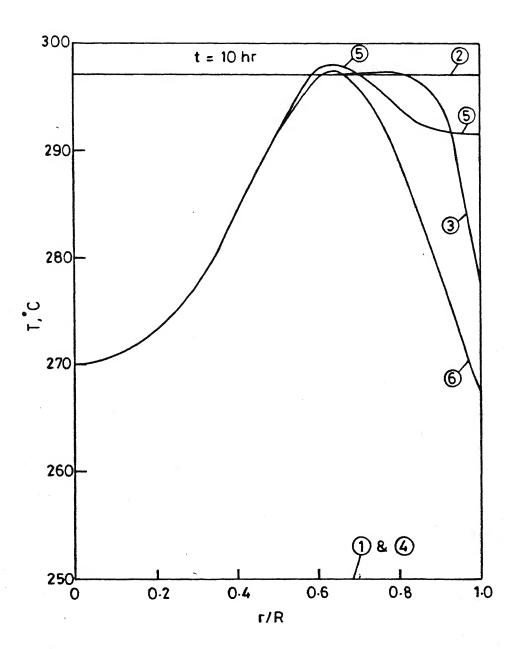


Fig. 6.

Radial profile for temperature at t=10 hr for various flow and heat transfer conditions (given in Table VII). Feed condition is as per eqn. 27.

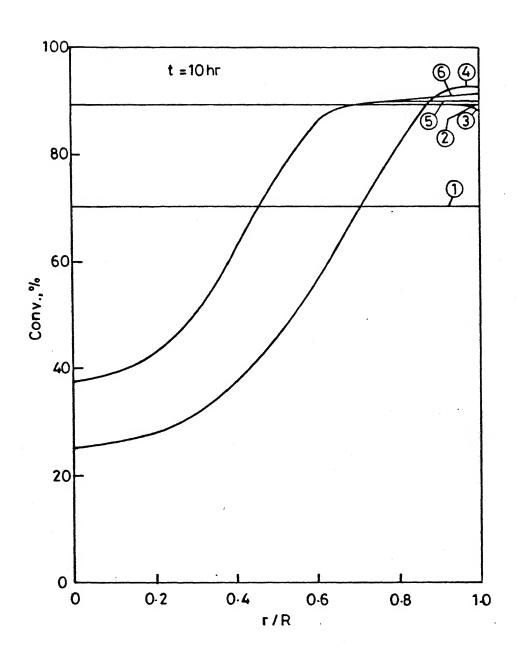


Fig. 7,

Radial profile for conversion at t=10 hr for various flow and heat transfer conditions (given in Table VII). Feed condition is as per eqn. 27.

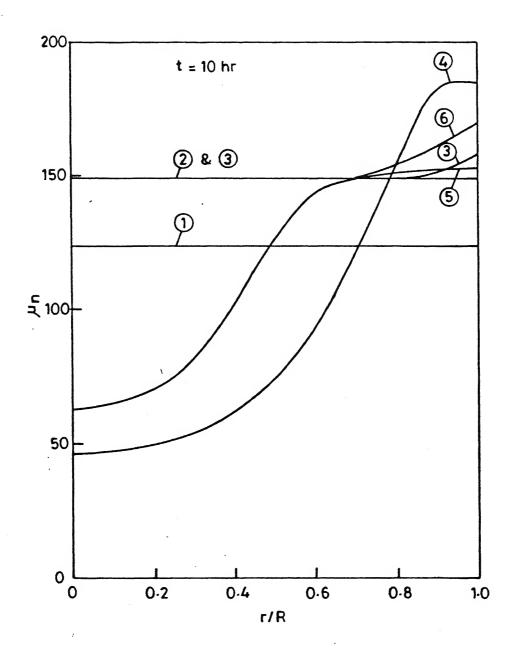


Fig.8.

Radial profile for M at t=10 hr for various flow and heat transfer conditions (given in Table VII). Feed condition is as per eqn. 27.

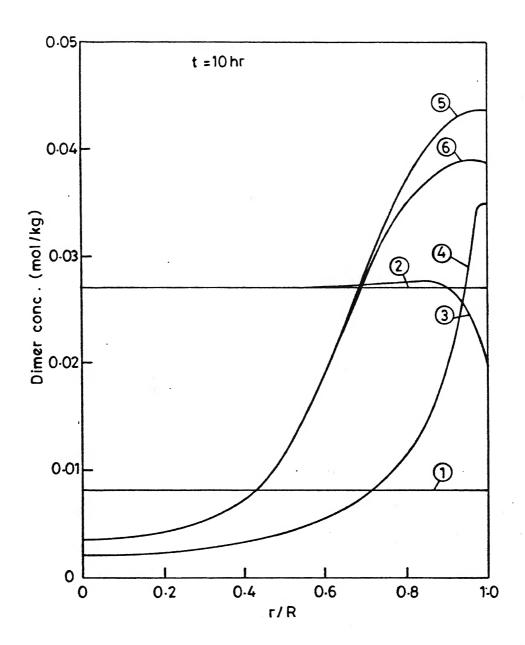


Fig. 9. Radial profile for  $\lfloor C_2 \rfloor$  at t=10 hr for various flow and heat transfer conditions (given in Table VII). Feed condition is as per eqn. 27.

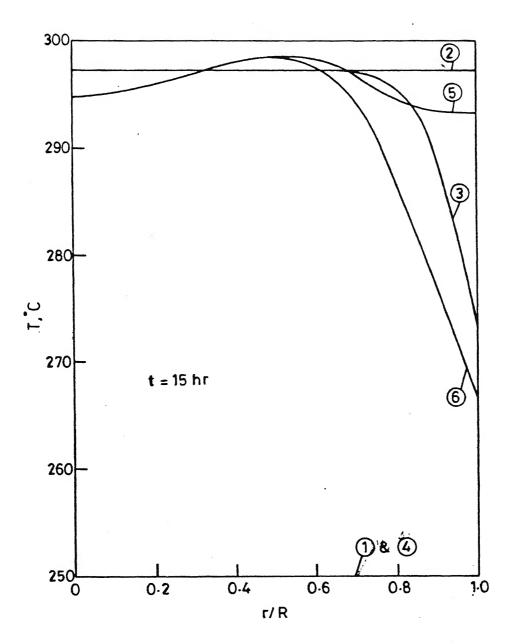


Fig. 10.

Radial profile for temperature at t=15 hr for various flow and heat transfer conditions (given in Table VII). Feed condition is as per eqn. 27.

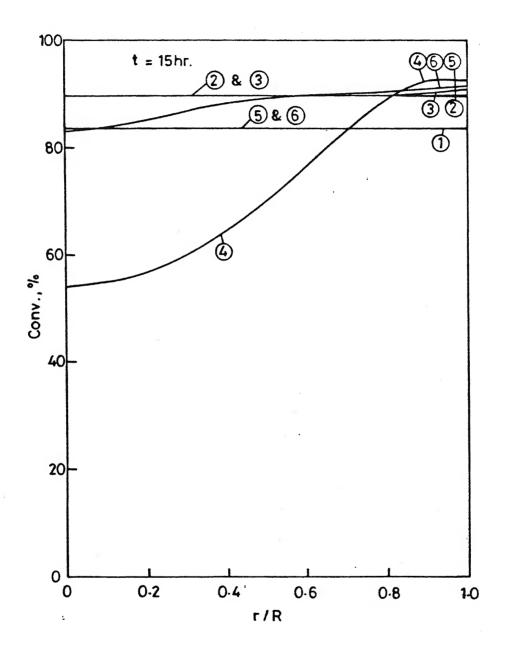


Fig.11.

Radial profile for conversion at t=15 hr for various flow and heat transfer conditions (given in Table VII). Feed condition is as per  $e_4n$ . 27.

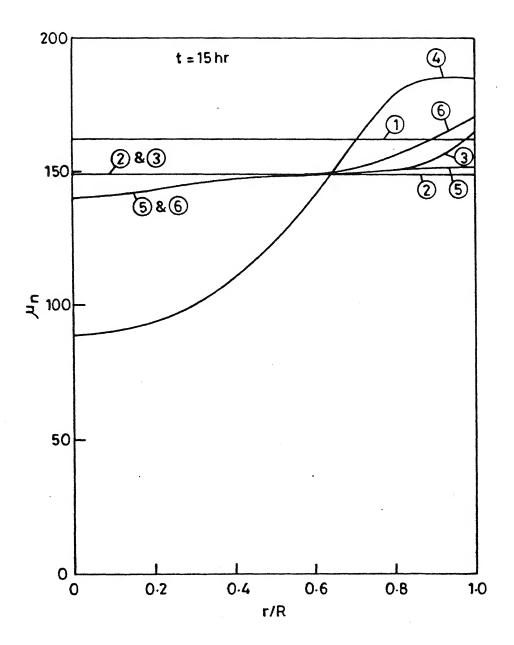


Fig.12.

Radial profile for  $\stackrel{\mathcal{M}}{n}$  at t=15 hr for various flow and heat transfer conditions (given in Table VII). Feed condition is as per eqn. 27.

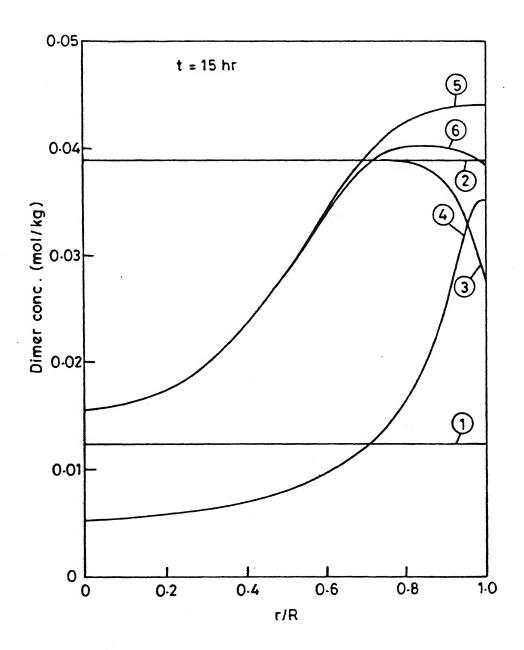


Fig. 13.

Radial profile for  $[C_2]$  at t=15 hr for various flow and heat transfer conditions (given in Table VII). Feed condition is as per eqn. 27.

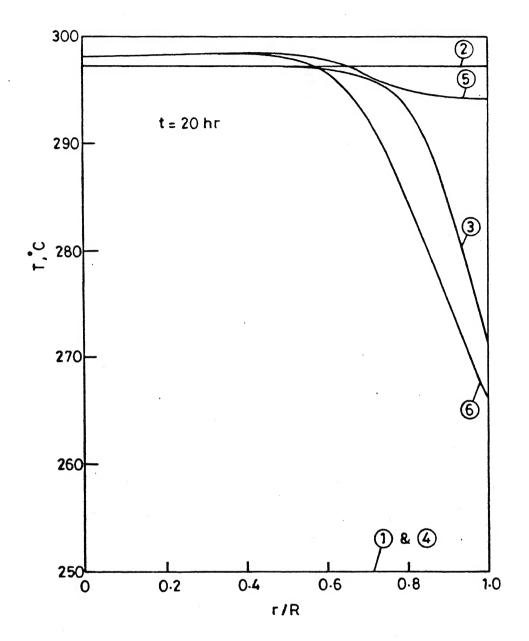


Fig.14.

Radial profile for temperature at t=20 hr for various flow and heat transfer conditions (given in Table VII). Feed condition is as per eqn. 27.

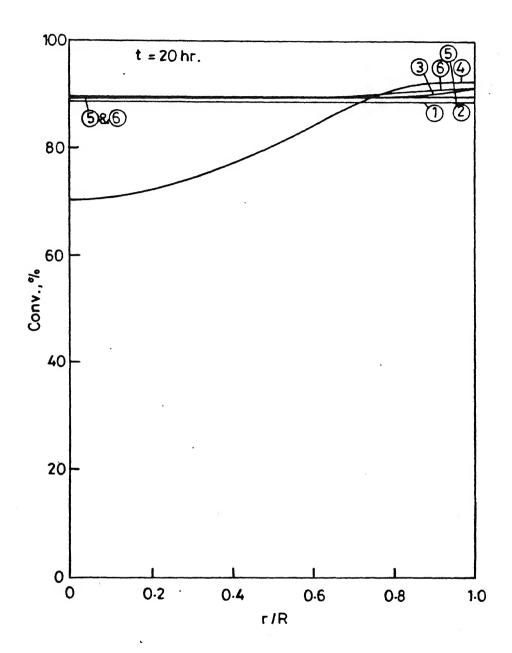


Fig.15.

Radial profile for conversion at t=20 hr for various flow and heat transfer conditions (given in Table VII). Feed condition is as per eqn. 27.

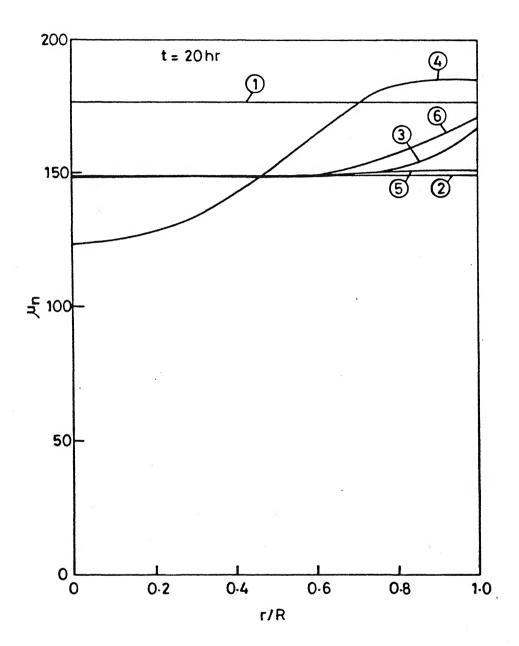


Fig.16.

Radial profile for  $\mu_n$  at t=20 hr for various flow and heat transfer conditions (given in Table VII). Feed condition is as per eqn. 27.

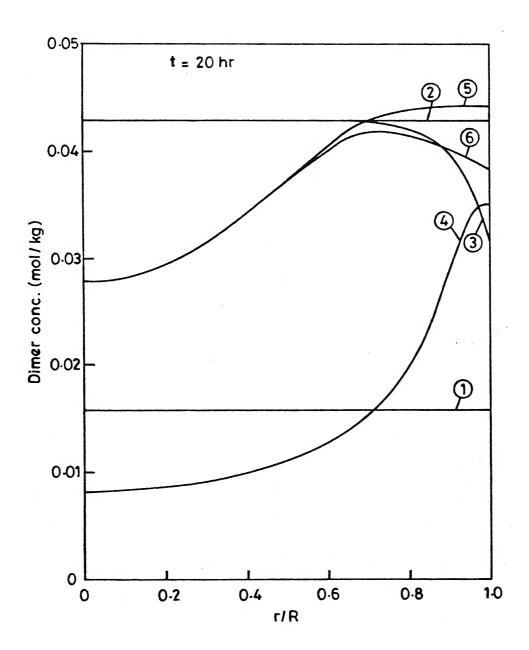
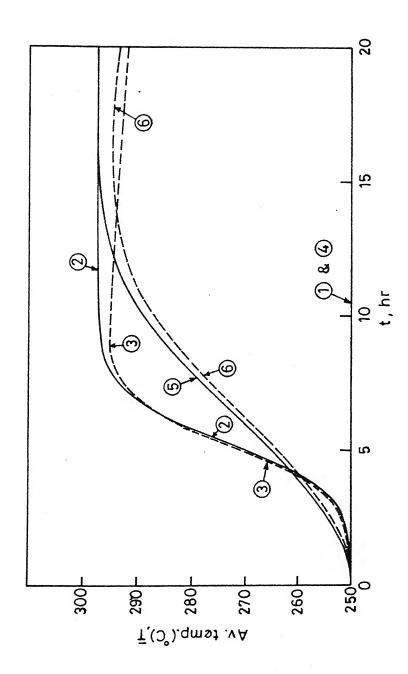


Fig.17. Radial profile for  $\lceil C_2 \rceil$  at t=20 hr for various flow and heat transfer conditions (given in table VII). Feed condition is as per eqn. 27.

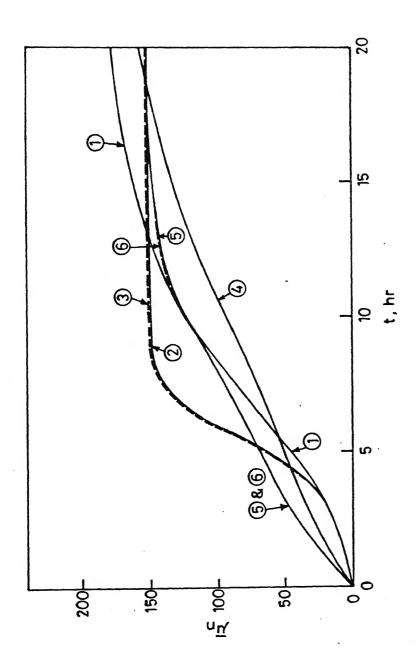
transfer effects. In the laminar adiabatic case (curve 5, Fig. 2), a temperature maximum first appears (at t near zero) at the wall because of almost instantaneous attainment of equilibrium conversions there. The higher temperature leads to thermal diffusion away from the wall as t increases (since h=0), contributing to a decrease in the temperature at the wall. In addition, conversion as t increases layers adjacent to the wall reach near-equilibrium/ (curves 5 & 6, Figs.3 to 5), generating more heat there. explains the temperature maximum in curve 5, Fig.2. The peak shifts further inwards as t increases (fig.6). Because of cooling from the jacket, the temperatures at the wall are lower for the laminar non-adiabatic case (curve 6, Fig.2). The slightly higher near-equilibrium conversions for curve 6, Fig.2 (compared to curve 5), is because of the lower temperatures present and the overall exothermic nature of the reactions. Since the equilibrium values of  $\mu_{n}$  are more sensitive to temperature, the difference between curves 5 and 6 (Fig.4) are greater. In plug flow, the residence times of fluid element at the wall are not infinite, and so, near-equilibrium conditions do not prevail at the wall for this ideal reactor. In the adiabatic case, the temperature (and other) profiles are flat as expected, since there is no driving force for creating radial differences. For the plug flow non-adiabatic case (curve 3, Fig.2) a temperature maximum near the wall develops again. Near t=0, a maximum appears at the wall  $(T_{,T} > T_{,O})$ . This beds to higher conversions at the wall, which simultaneously leads to higher heat generation by reaction at that point. Very soon the temperature

at the wall increases beyond  $T_J$ . Heat transfer takes place both to the jacket as well as towards the center. The conversion is the highest at the wall for case 3 (Fig. 3) since it represents a Cumulative effect from t=0 to 5. Figs. 6 to 17 show conditions at t=10, 15 and 20 hr. The trends at t=5 hr continue, but some flattening of temperature profiles in observed, because of thermal diffusion effects away from the peak, both towards the wall as well as towards the center. The conversion,  $\mu_n$  and cyclic dimer profiles show higher levels near the center as t increases, as expected. Figs. 15 and 17, at t=20hr, show that even though the monomer conversion has almost attained equilibrium values throughout the cross-section, lower cyclic dimer concentrations near the center are observed. This is because equilibrium for the cyclic dimer reactions has not yet been attained.

Figs. 18 and 19 show the profiles of average temperature,  $\overline{T}$ , and average chain length,  $\overline{\mu}_n$ , along the reactor length for various flow and heat transfer conditions. It is interesting to note that the distinct "S' - shape for the plug flow case (curves 2 & 3, Fig.18) is relatively less prominent in the laminar flow case (curves 4 & 5, Fig.18). Average temperatures for the non-adiabatic ease (curves 3 & 6, Fig.18) are in general lower than that for the adiabatic case (curves 2 & 3, Fig.18) because of the cooling effects of the jacket fluid, except for the initial part of the plug flow case (curves 2 & 3) where the jacket fluid has a heating effect as discussed earlier. For  $\overline{\mu}_n$ , again, the



Axial variation of  $\bar{T}$  for various flow and heat transfer conditions (given in Table VII) with feed condition as per eqn. 27.



conditions (given in Table VII) with feed condition as pereqn. 27. Axial variation of  $\widetilde{\mathcal{M}}_n$  for various flow and heat transfer

Fig.19.

'S' - shape in case of plug flow (curves 1,2 & 3, Fig.19) is less prominent in laminar flow (curves 4,5 & 6 Fig.19). In general, isothermal operation (curves 1 & 4, Fig.19) gives lower  $\overline{\mu}_n$  in the initial part of the reactor because of comparatively lower temperatures in this case. But as t increases and equilibrium conversions are attained,  $\overline{\mu}_n$  for the isothermal case registers higher values.

Table VIII shows the effects of the various parameters, e.g., radius of the reactor (R), coolant temperature ( $T_J$ ), feed temperature ( $T_0$ ) and inlet water concentration ( $[w]_0$ ), on the average values of temperature, conversion,  $\boldsymbol{\mu}_{\,\boldsymbol{n}}$  and cyclic dimer concentration. Decreasing the value of R to 0.4 (Run-2) from the reference conditions (Run-1) helps increase radial heat transfer. Thus, the initial heating effects of the jacket fluid increases T at 5 hrs over that for the reference run. As t increases, the temperature is comparatively lower due to better cooling. However, average conversion and  $\overline{\mu}_n$  are always lower. Increasing the value of R (Run-3, Table VIII) has the opposite effects. A lower coolant temperature (Run-4, Table VIII) causes lower temperature in the reactor. This reduces the average conversion and  $\overline{\mu}_n$  initially. But, as equilibrium is attained at higher values of t, the average conversion and  $\overline{\mu}_{n}$ , are higher as an effect of lower temperatures. A lower feed temperature (Run-5, Table VIII) causes lower T. As before, average conversion and  $\overline{\mu}_n$  are initially lower, but

Table VIII

Acc. No. A.104091

## Effect of Various Parameters on Average Values

Run	Description	t(hr)	T A	wg.Conv.	,% µ n	[ ट <sub>2</sub> ]
1.	Reference Condition:	5	264.09	28.24	67.08	0.00516
	Laminar flow, Non-adiabatic (h=5)	10	287.20	73.66	123.83	0.01788
	$R = 0.6m$ $T_{T} = 260^{\circ}C$	15	294.09	88.50	149.61	0.03003
T <sup>J</sup> = Feed	T <sup>J</sup> = 250°C Feed condition as per eqn.27	20	293.83	89.65	151.36	0.03702
2.	$R^a = 0.4m$	5	264.27	<b>2</b> 9.09	68.79	0.00497
		10	287.06	74.89	125.46	0.01782
		15	292.90	88.80	150.72	0.03002
		20	292.01	89.78	152.52	0.03684
3.	$R^a = 0.8m$	5	263.96	27.66	65,69	0.00524
		10	287.39	73.19	122.80	0.01786
		15	294.71	88.38	149.09	0.03000
		20	294.73	89.59	150.80	0.03706
4.	$T_J^a = 250$ °C	5	263.49	28.03	66.71	<b>0.004</b> 86
		10	286.44	73.67	124.19	0.01751
		15	293.12	88.55	150.22	0.02969
		20	292.64	89.73	152.13	0.03670
5.	$T_0^a = 240$ °C	5	247.96	15.38	53.06	0.00253
		10	264.57	47.73	89.55	0.00940
	***	15	278.46	74.90	128.58	0.01757
		20	284.75	88.09	153.77	0.02685

Table VIII (Contd.)

6.	$[W]_0^a = 0.13 \text{ moles/kg}$	5	260.25	20.76	65.07	0.00352
		10	280.21	60.04	117.87	0.01301
		15	292.05	84.16	161.87	0.02413
		20	293.84	89.17	170.77	0.03260
7.	$[W]_{o}^{a} = 0.18 \text{ moles/kg}$	5	266.66	33.24	67.38	0.00638
		10	290.52	80.16	125.50	0.02108
		15	294.38	89.30	140.50	0.03305
		20	293.72	89.71	141.25	0.03892

a All other value as in reference condition.

are higher near equilibrium. Lower feed water concentration (Run.6, Table VIII) causes lower conversion and  $\overline{\mu}_n$  initially, as the ring opening step (step.1, Table I) is slower. But as t increases, and near-equilibrium conditions are attained, conversion improves and reaches the values for the reference state, since equilibrium conversion of polycondensation reaction (step.2, Table I) is higher for lower [W]. But the effect on  $\overline{\mu}_n$  is much more pronounced. This is because a shift of the equilibrium for the polycondensation step to the right hand side increases the chain length considerably. The effects of higher initial water concentration (Run.7, Table VIII), are opposite in comparison to run.6, Table VIII.

The effect of some 'computational variables' is now studied. The values of the 'slip', i.e., the non-dimensional velocity ( $v_z$  (r=R)/ $v_{av}$ ) assumed at the wall(to avoid overflow and other computational problems) was changed aroung the reference value of  $10^{-4}$  and the effects are shown in Table IX. It is observed that the results are relatively unaffected by changing the slip, but the CPU time requirement increases for lower values of slip. Similarly, instead of using the actual feed conditions at t=0, r=R, values of  $\begin{bmatrix} C_1 \\ 0 \end{bmatrix}$  ( $\begin{bmatrix} W \\ 0 \end{bmatrix}$ ),  $\begin{bmatrix} P_1 \\ 0 \end{bmatrix}$ , etc., close to equilibrium values at the wall temperature were used for the laminar flow case, to makes the equations less stiff. These values were changed to see their effect on the numerical results further down the reactor. The results were found to be quite

Table IX

Effect of Slip on Simulation Runs

Run <sup>a</sup>	Slip Used	Wall Temp. at 5 hr (°C)	CPU Time (for t= 5 hr)
1	10 <sup>-4</sup> (ref)	287.51	2 min 25 sec
2	10 <sup>-6</sup>	287.51	3 min 5 sec
3	10-3	287.51	2 min 6 sec

<sup>&</sup>lt;sup>a</sup>All runs are for laminar flow adiabatic reactor with feed condition as per eqn. 27

insensitive, but feeding in the near-equilibrium values reduced the computational time.

points on the results at t=5 hrs. The runs were made using method III, with the spacing of grid points denser near the wall (eqn.16). The figure shows very clearly that the results, particularly near the wall, converge as the number of points are increased and after about 8 points the results do not show any major discrepancy.

Methods I and III (for identical placement of FD points)were found to give the same results for various reactor conditions. This confirms that these two computer programs, one in the NAG Library and one made by us, are essentially the same. Both of them use the FD technique with GEAR's method. Method III, however, is more flexible since it permits the use of unequal spacing of FD grid points.

Some discrepancy near the wall was found between methods I and II. Table X shows the wall temperatures (which is the most sensitive variable ) for the laminar flow adiabatic reactor as predicted by methods I and II, at t=5 hrs. With 9FD grid points, the results show some differences at the wall, though those at the internal positions agree well. The reason for this discrepancy lies in the fact that in method II, the wall temperature is obtained using the boundary condition alone (eqn.15). Therefore

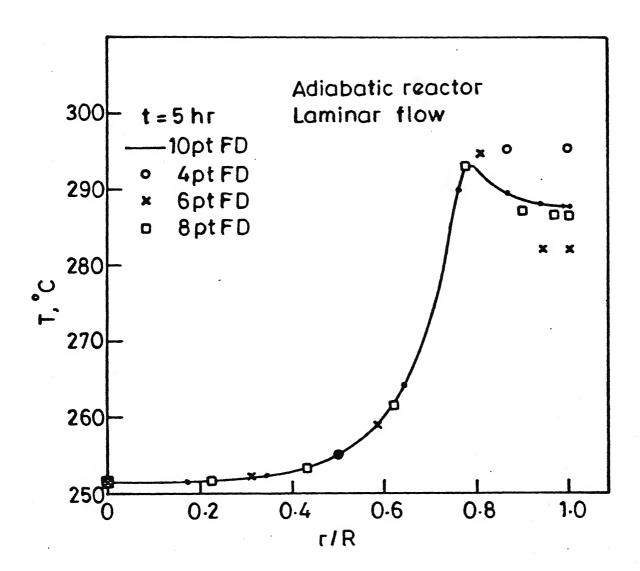


Fig. 20.

Temperature profile at t=5 hr in a laminar flow adiabatic reactor (feed conditions given in eqn. 27) for different number of grid points using finite difference (method III).

Table X

Comparison of Methods I and II

(Laminar Flow Adiabatic Reactor)

No. of	Temp. at the Wall	(T <sub>N+1</sub> ), °C, at t=5hr		
FD pts., — (N+1)	Method-I	Method-II		
9	290.04	294.85		
12	285.72	288.68		
15	287 <b>.62</b>	288.04		
20	287.50	287.50		
*				

the ODE from the heat balance equation at the wall (r=R), and thus the heat of reaction information near that location, is not used. In method I, on the other hand, both the boundary condition and the ODE from the heat balance are used at r=R. Therefore, when the gradients are steep near the wall (for lower values of t), method I gives better predictions. Thus method I is superior to method II, particularly with regard to the results near the wall. However, as the numbers of FD points are increased, the results for both the method converge as seen in Table X. Similar results are obtained for the non-adiabatic case. However, for the non-adiabatic case of the plug flow reactor, where the temperature gradients near the wall are not so steep, the results agree well for the two methods (I & II).

The orthogonal collocation technique gives results which match those from method III for all cases of plug flow reactor (run. 1-3, Table VII) as also for the isothermal laminar flow reactor (run.4, Table VII). However, the predictions do not match well for the adiabatic (see Fig.21) and non-adiabatic cases of the laminar flow reactor (Runs. 2 & 3, Table VII). The mismatch is particularly large for lower values of t, when a steep gradient of temperature exists near the wall. The ability of our computer program for the OC method to predict good results for cases where the temperature gradients are not too steep near the wall, indicates the correctness of the program.

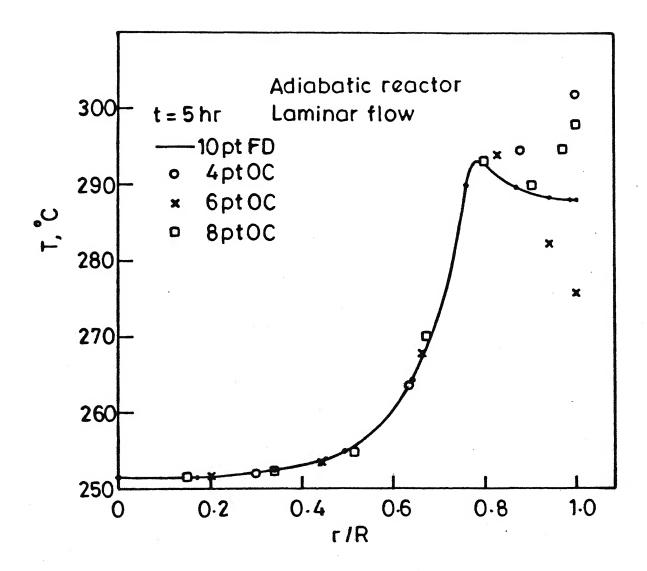


Fig. 21.

Temperature profile at t=5 hr in a laminar flow adiabatic reactor (feed condition given by eqn. 27) for different number of grid points using orthogonal collocation (method IV)

It is felt that the incorrect results obtained for the non-isothermal cases of the laminar flow reactor at lower values of t could be attributed to the fact that like method II, the OC method also does not incorporate the ODE from the heat balance at the wall. Instead, it computes the wall temperature  $(T_{N+1})$  using all the other values of temperature,  $T_1$ ,  $T_2$ ,...,  $T_N$ (eqn.20). The OC method does not compare too well with method II either. This is due to the fact that in calculating the wall temperature ( $T_{N+1}$ ), method II uses information only for the two adjacent points,  $\mathbf{T}_{N}$  and  $\mathbf{T}_{N-1}$  (eqn.15), whereas the OC method requires information of all other points (eqn.20), giving different weightages to them. Thus the prediction of the wall temperature in the OC method is likely to be sensitive to the position of the maximum (as it shifts) in the temperature profile. As t increases, the temperature maximum flattens out and the OC method is indeed found to give better results. The results for the OC method, where steep temperature gradients exist, are likely to improve on the introduction of the finite element technique. 23

## CONCLUSIONS

A detailed simulation of nylon-6 polymerization in continuous flow tubular reactors was carried out using four methods. Two types of finite difference techniques as well as the method of orthogonal collocation were used to convert PDEs into ODEs. GEAR's method was used to solve the resultant set of ODEs.

The simulation results showed the existence of significant radial gradients for temperature and concentrations for the laminar flow cases, particularly for lower values of t (average residence time). A temperature peak forms at lower t near the wall, which flattens out and simultaneously shifts towards the center further downstream. The monomer conversion and number average chain length profiles result from the temperature profiles. Higher temperature normally increases conversion and number average chain length, but it has a reverse effect where near-equilibrium conditions exist. A characteristic 'S'shape was observed in case of plug flow in the plot of average temperature and  $\overline{\mu}_n$  versus t. The plot was much flatter in the laminar flow case. The cases of adiabatic and non-adiabatic reactors both with plug and laminar flow were found not to give much different results for the average value of  $\mu_n$ , which is of greater importance as far as the final product properties are concerned.

Methods I and III were found to be essentially similar with the latter having more flexibility as it could use any type of spacing of grid points. Method II was found to give poorer results near the wall particularly when the temperature gradient near the wall was steep. Both the methods I and II agreed wall as the number of grid points was increased. Method IV gave good results for all the cases except the adiabatic and non-adiabatic cases in laminar flow situation near the wall where the temperature gradients were particularly steep. Use of the finite element method is suggested to give improved results for such cases.

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```
****************
FINITE DIFFERENCE: METHOD I
                         *******************
                        ICOL=NPTS=1
IF(ICOND.E0.0) WRITE(39,17)
IF(ICOND.E0.1) WRITE(39,18)
IF(ICOND.E0.2) WRITE(39,19)
FORMAT(15X,50('*'),///,30X,'ISOTHERMAL REACTOR',///,15X,50('*')-
17
                         FORMAT(15x,50('*'),///,30x, 'ADIABATIC REACTOR',///,15x,50('*'),
18
                        FORMAT(15x,50('*'),//,30x, ADIABATIC REACTOR',//,15x

1 //)

FORMAT(15x,50('*'),//,22x, TUBULAR REACTOR WITH HEAT

1 TRANSFER',//,15x,50('*'),//)

WRITE(39,20) R,NPTS,U10,U40,U20,U50,U70,U30,U60

FORMAT(5x, RADIUS=',F6.2,5x, NO OF F.D. POINTS=',I3,/,

1 5x, MONOMER=',F8.4,5x,'1st MOMENT=',F8.4,5x,/,

2 8x,'P(1)=',F8.4,10x,'DIMER=',F8.4,10x,'TEMP=',F10.3

2x,'Oth MOMENT=',F8.4,10x,'WATER=',F8.4,/,75('*'))

TN=TW-273.

IF(ICOND.E0.2) WRITE(39,21) UO,TN

FORMAT(/,5x,'HEAT TRANSF COEFF=',F10.5,2x,'KCAL/(SO M 1 (DEG K)',/,10x,'COULANT TEMP=',F10.5,//,7S('*'))

TOL=1.E=06

IRELAB=2

MPED=0

X=0.0
19
20
21
                        X=0.0
H=0.5
IT=0
                        IT=0
DR=1.0/ICOL
DO 11 I=1.(ICOL+1)
COL(I)=FLOAT(I-1)*DR
DO 33 J=1.ICOL+1
RX(J)=2.*(1.-(COL(J))**2.)
IF(RX(J).EQ.O.O) RX(J)=1.E-04
IF(IVEL.EQ.O) RX(J)=1.0
CONTINUE
INITIAL CONDITION
11
33
C
                        CONTINUE
INITIAL CONDITION
DO 22 I=1,(ICOL+1)
Y(I)=U10
Y(ICOL+1+I)=U20
Y(ICOL+2+2+I)=U30
                         Y(ICOL*3+3+1)=U40
Y(ICOL*4+4+1)=U50
Y(ICOL*5+5+1)=U60
Y(ICOL*6+6+1)=U70
CONTINUE
                        CONTINUE

IF(IVEL.EQ.0) GO TO 23

Y(ICOL+1)=0.670647

Y(ICOL*2+2)=0.000234

Y(ICOL*3+3)=0.043585

Y(ICOL*4+4)=8.05944

Y(ICOL*5+5)=0.034956

Y(ICOL*6+6)=0.116415

IFAIL=0

N=(ICOL+1)*7

IW=18+N

CALL DOZEBF(X.XEND.N.Y
22
23
                          CALL DOZEBF(X, XEND, N, Y, TOL, IRELAB, FCN, MPED,
```

```
PEDERV, OUTPUT, W, IW, IFAIL)
WRITE(39,26) (COL(I), I=1, ICOL+1)
FORMAT(10X, 10F8.3)
IF(IFAIL.NE.0) GO TO 24
STOP
TYPE *, IFAIL
26
24
     CCC
44
C
16
     C
11
```

```
F(J)=-FK(1,J)*Y(J)*Y(1cot*5+5+J)+BK(1,J)*Y(1cot+1+J)*
1 -FK(3,J)*Y(J)*Y(ICOt*2+2+J)
2 +BK(3,J)*(Y(ICOt*2+2+J)-Y(ICOt+1+J))
F(J)=F(J)/RX(J)
C
                          f(icoL+i+j)=fK(i,j)*Y(j)*Y(icoL*5+5+j)-BK(i,j)*Y(icoL+1+J)

1 -2.*FK(2,J)*Y(icoL+1+J)*Y(icoL*2+2+J)

2 +2.*BK(2,J)*Y(icoL*5+5+J)*(Y(icoL*2+2+J)-Y(icoL+1+J))

3 -FK(3,J)*Y(icoL+1+J)*Y(J)+

BK(3,J)*P2-FK(5,J)*Y(icoL+1+J)*Y(icoL*4+4+J)+BK(5,J)*P3
F(icoL+1+J)=F(icoL+1+J)/RX(J)
C
                          F(ICOL*2+2+J)=FK(I;J)*Y(J)*Y(ICOL*5+5+J)-BK(I;J)*Y(ICOL+1+J)

-FK(2,J)*Y(ICOL*2+2+J)*Y(ICOL*2+2+J)+

BK(2,J)*Y(ICOL*5+5+J)*(Y(ICOL*3+3+J)-Y(ICOL*2+2+J))

+FK(4,J)*Y(ICOL*4+4+J)*Y(ICOL*5+5+J)-

BK(4,J)*P2

F(ICOL*2+2+J)=F(ICOL*2+2+J)/RX(J)
C
                          F(ICOL*3+3+J)=FK(1,J)*Y(J)*Y(ICOL*5+5+J)-BK(1,J)*Y(ICOL+1+J)
1 +FK(3,J)*Y(J)*Y(ICOL*2+2+J)-
2 BK(3,J)*(Y(ICOL*2+2+J)-Y(ICOL+1+J))+2.*FK(5,J)*
3 Y(ICOL*4+4+J)*Y(ICOL*2+2+J)-2.*BK(5,J)*
4 APR+2.*FK(4,J)*Y(ICOL*5+5+J)*Y(ICOL*4+4+J)-2.*BK(4,J)*P2
F(ICOL*3+3+J)=F(ICOL*3+3+J)/RX(J)
C
C
                          f(icoL*4+4+J)==fk(4,J)*Y(icoL*4+4+J)*Y(icoL*5+5+J)+Bk(4,J)*P2
1 -FK(5,J)*Y(ICOL*4+4+J)*Y(ICOL*2+2+J)+
2 BK(5,J)*APR
F(ICOL*4+4+J)=F(ICOL*4+4+J)/RX(J)
C
                           F(icoL*5+5+J)=-FR(i,j)*Y(J)*Y(icoL*5+5+J)+BR(i,J)*Y(icoL+1+J)

+FR(2,J)*Y(ICOL*2+2+J)*Y(ICOL*2+2+J)-

BR(2,J)*Y(ICOL*5+5+J)*(Y(ICOL*3+3+J)-Y(ICOL*2+2+J))

-FR(4,J)*Y(ICOL*4+4+J)*Y(ICOL*5+5+J)+BR(4,J)
                           2 *P2
F(ICOL*5+5+J)=F(ICOL*5+5+J)/RX(J)
                         F(ICOL*5+5+J)=F(ICOL*5+5+J)/RX(J)

continue
D0 13 JJ=1.ICOL+1
P2=Y(ICOL+1+JJ)
P3=Y(ICOL+1+JJ)
APR=(Y(ICOL*2+2+JJ)-2.*Y(ICOL+1+JJ))
IF(APR.LT.0.0) APR=0.0
IF(APR.LT.0.0) P3=0.0
IF(APR.LT.0.0) P3=0.0
F1(JJ)=-DH(1)*(FK(1,JJ)*Y(JJ)*Y(ICOL*5+5+JJ)-BK(1,JJ)*
7 Y(ICOL+1+JJ)
1 -DH(2)*(FK(2,JJ)*Y(ICOL*2+2+JJ)*Y(ICOL*2+2+JJ)-BK(2,JJ)*
2 Y(ICOL*5+5+JJ)*(Y(ICOL*3+3+JJ)-Y(ICOL*2+2+JJ)-BK(3,JJ)*
4 Y(ICOL*2+2+JJ)+BK(3,JJ)*Y(ICOL*2+2+JJ)-BK(3,JJ)*
4 Y(ICOL*2+2+JJ)+BK(3,JJ)*Y(ICOL*4+4+JJ)*Y(ICOL*5+5+JJ)-BK(4,JJ)*P21
CONTINUE
C
12
13
C
                           CONTINUE
                           DO 15 L=2, ICOL+1
F2(L)=(TK/R**2.0)*((TP(L+1)-TP(L-1))/2.0/DR
1 /COL(L)+(TP(L+1)-2.0*TP(L)+TP(L-1))/(DR**2.0))
                           CONTINUE

TYPE *,F1(ICOL+1),F2(ICOL+1)

F2(1)=4.0*TK/(R**2.0)*(TP(2)-TP(1))

DO 17 J=1,ICOL+1

F(ICOL*6+6+J)=(F2(J)+RHO(J)*F1(J)/1000.0)
15
C
                            2 /(RX(J)*RHO(J)*CP(J))
IF(ICOND.EQ.O) F(ICOL*6+6+J)=0.0
CONTINUE
 17
                            RETURN
                            C
```

```
C
                                                                                          SUBROUTINE OUTPUT(XSOL,Y)
DIMENSION Y(150),RHO(25),CP(25),
1 COL(25),RY(10,25,50),TY(15,50),U(25),UU(25),U1(25),U2(25),
2 U3(25),U4(25),U5(25),U6(25),U7(25),RX(25),TP(25)
COMMON /OUT/H,IT,X,N
COMMON ICOL,COL,U10,RX,U0,TW,R,TK,U70,ICOND,IVEL,DR
                                                                                              SUBROUTINE
                                                                                  COLITIONS
COMMON COLLOG., 101, 123, 16(25), U7(25), E1(25), U1(25), U1
55
17
16
18
 22
  39
  23
  24
  26
```

```
*******************
FINITE DIFFERENCE: METHOD II
                      ****************
                      ICOL=NPTS-1
IF(ICOND.EO.0) WRITE(39,17)
IF(ICOND.EO.1) WRITE(39,18)
IF(ICOND.EO.2) WRITE(39,19)
FORMAT(15X,50('*'),//,30X,'ISOTHERMAL REACTOR',///,15X,50('*').
17
                       FORMAT(15x,50('*'),///,30x,'ADIABATIC REACTUR',///,15x,50('*'),
18
                      FORMAT(15X,50('*'),//,30X, 'ADIABATIC REACTUR',//,15X,50('')

FORMAT(15X,50('*'),//,22X, 'TÜBÜLAR REACTOR WITH HEAT

1 TRANSFER',//,15X,50('*'),//)

WRITE(39,20) R, HPTS, U10, U40, U20, U50, U70, U30, U60

FORMAT(5X, 'RADIUS=',F6.2,5X,'NO OF F.D. POINTS=',I3,/,

5X, 'MONOMER=',F8.4,5X,'1st MOMENT=',F8.4,5X,/,

8X,'P(1)=',F8.4,10X,'DIMER=',F8.4,10X,'TEMP=',F10.4,/,

3 2X,'Oth MOMENT=',F8.4,10X,'WATER=',F8.4,/,75('*'))

TN=TW-273.

IF(ICOND.E0.2) WRITE(39,21) U0,TN

FORMAT(/,5X,'HEAT TRANSF COEFF=',F10.5,2X,'KCAL/(SO M)(HR))

1 (DEG K)',/,10X,'COOLANT TEMP=',F10.5,//,75('*'))

TUL=1.E-06

IRELAB=2

MPED=0
19
20
21
                      MPED=0
                      X=0.0
H=0.5
IT=0
                      TT=0
DR=1.0/ICOL
DO 11 I=1.(ICOL+1)
COL(I)=FLOAT(I-1)*DR
DO 33 J=1.ICOL+1
RX(J)=2.*(1.-(COL(J))**2.)
IF(RX(J).EQ.0) RX(J)=1.E-04
IF(IVEL.EQ.0) RX(J)=1.0
11
33
C
                        CONTINUE
                       INITIAL CONDITION
DO 22 I=1,(ICOL+1)
Y(I)=U10
Y(ICOL+1+I)=U20
Y(ICOL+2+2+I)=U30
                      Y(ICOL*3+3+1)=U40
Y(ICOL*4+4+1)=U50
Y(ICOL*5+5+1)=U60
Y(ICOL*6+6+1)=U70
CONTINUE
22
                       CUNTINGE

IF (IVEL.EQ.0) GO TO 23

Y(ICOL+1)=0.670647

Y(ICOL*2+2)=0.000234

Y(ICOL*3+3)=0.043585

Y(ICOL*4+4)=8.05944

Y(ICOL*5+5)=0.034956

Y(ICOL*6+6)=0.116415
                       IFAIL=0
N=(ICOL+1)*7-2
23
                       IW=18+N
CALL DOZEBF(X, XEND, N, Y, TOL, IRELAB, FCN, MPED,
```

```
PEDERV, OUTPUT, W, IW, IFAIL)
WRITE(39,26) (COL(I), I=1, ICOL+1)
FORMAT(10X, 20F8.3)
IF(IFAIL.NE.0) GO TO 24
STOP
TYPE *, IFAIL
END...
26
24
                          CCC
                       44
C
16
                        DS(5)=5.8265E-01

AC(1)=4.3075E+07
AC(2)=1.2114E+10
AC(3)=1.6377E+10
AC(4)=2.3307E+12
AC(5)=3.011E+09
EC(1)=1.8866E+04
EC(2)=2.067E+04
EC(3)=2.0107E+04
EC(4)=3.74E+04
EC(5)=2.04E+04
UR=1.98
DO 11 J=1,ICOL+1
FK(I,J)=AD(I)*EXP(-EO(I)/(UR*TP(J)))+AC(I)*EXP
1 (-EC(I)/(UR*TP(J)))*Y(ICOL*2+2+J)
EK(I,J)=EXP(DS(I)/UR-DH(I)/(UR*TP(J)))
BK(I,J)=FK(I,J)/EK(I,J)
CONTINUE
DO 12 J=1,ICOL+1
P2=Y(ICOL+1+J)
P3=Y(ICOL+1+J)
APR=(Y(ICOL+1+J)
APR=(Y(ICOL*2+2+J)-2.*Y(ICOL+1+J))
IF(APR.LT.0.0) APR=0.0
C
11
```

```
IF(APR.LT.0.0) P2=0.0
IF(APR.LT.0.0) P3=0.0
C
                            F(J)=-FK(1,J)*Y(J)*Y(ICOL*5+5+J)+BK(1,J)*Y(ICOL+1+J)*
1 -FK(3,J)*Y(J)*Y(ICOL*2+2+J)
2 +BK(3,J)*(Y(ICOL*2+2+J)-Y(ICOL+1+J))
F(J)=F(J)/RX(J)
                           f(icoL+i+j)=fK(i,j)*Y(j)*Y(icoL*5+5+j)-BK(i,j)*Y(icoL+1+J)

1 -2.*FK(2,J)*Y(icoL+1+J)*Y(icoL*2+2+J)

2 +2.*BK(2,J)*Y(icoL*5+5+J)*(Y(icoL*2+2+J)-Y(icoL+1+J))

3 -FK(3,J)*Y(icoL+1+J)*Y(J)+

BK(3,J)*P2-FK(5,J)*Y(icoL+1+J)*Y(icoL*4+4+J)+BK(5,J)*P3

F(icoL+1+J)=F(icoL+1+J)/RX(J)
C
C
                           F(icoL*2+2+j)=FK(i,j)*Y(j)*Y(icoL*5+5+j)-BK(i,j)*Y(icoL+1+J)

-FK(2,J)*Y(icoL*2+2+J)*Y(icoL*2+2+J)+

BK(2,J)*Y(icoL*5+5+J)*(Y(icoL*3+3+J)-Y(icoL*2+2+J))

+FK(4,J)*Y(icoL*4+4+J)*Y(icoL*5+5+J)-

BK(4,J)*P2
F(icoL*2+2+J)=F(icoL*2+2+J)/RX(J)
                           F(İCOL*3+3+J)=FR(i,J)*Y(J)*Y(İCOL*5+5+J)-BR(i,J)*Y(İCOL+1+J)

1 +FK(3,J)*Y(J)*Y(ICOL*2+2+J)-

2 BK(3,J)*(Y(ICOL*2+2+J)-Y(ICOL+1+J))+2.*FK(5,J)*

3 Y(ICOL*4+4+J)*Y(ICOL*2+2+J)-2.*BR(5,J)*

4 APR+2.*FK(4,J)*Y(ICOL*5+5+J)*Y(ICOL*4+4+J)-2.*BK(4,J)*P2

F(ICOL*3+3+J)=F(İCOL*3+3+J)/RX(J)
C
C
                            F(icoi*4+4+j)=-FK(4,j)*Y(icoi*4+4+j)*Y(icoi*5+5+j)+BK(4,j)*P2
1 -FK(5,j)*Y(icoi*4+4+j)*Y(icoi*2+2+j)+
BK(5,j)*APR
                            2 BK(5,J)*APR
F(ICOL*4+4+J)=F(ICOL*4+4+J)/RX(J)
                            F(icoi*5+5+j)=-FK(i;j)*Y(j)*Y(icoi*5+5+j)+BK(i;j)*Y(icoi+1+J)

+FK(2,J)*Y(ICOL*2+2+J)*Y(ICOI*2+2+J)-

BK(2,J)*Y(ICOI*5+5+J)*(Y(ICOI*3+3+J)-Y(ICOI*2+2+J))

-FK(4,J)*Y(ICOI*4+4+J)*Y(ICOI*5+5+J)+BK(4,J)
                            2 *P2
F(ICOL*5+5+J)=F(ICOL*5+5+J)/RX(J)
                           CONTINUE
DO 13 JJ=2,ICOL
P2=Y(ICOL+1+JJ)
P3=Y(ICOL+1+JJ)
APR=(Y(ICOL*2+2+JJ)-2.*Y(ICOL+1+JJ))
IF(APR.LT.0.0) APR=0.0
IF(APR.LT.0.0) P2=0.0
IF(APR.LT.0.0) P3=0.0
F1(JJ)=-DH(1)*(FK(1,JJ)*Y(JJ)*Y(ICOL*5+5+JJ)-BK(1,JJ)*
7 Y(ICOL+1+JJ))
C
12
                                     JJ)=-DH(1)*(FK(1,00)*1(00,+1(10))

Y(ICOL+1+JJ))

-DH(2)*(FK(2,JJ)*Y(ICOL*2+2+JJ)*Y(1COL*2+2+JJ)-BK(2,JJ)*

Y(ICOL*5+5+JJ)*(Y(ICOL*3+3+JJ)-Y(ICOL*2+2+JJ)))

-DH(3)*(FK(3,JJ)*Y(JJ)*Y(ICOL*2+2+JJ)-BK(3,JJ)*

Y(ICOL*2+2+JJ)+BK(3,JJ)*Y(ICOL*1+JJ))

-DH(4)*(FK(4,JJ)*Y(ICOL*4+4+JJ)*Y(ICOL*5+5+JJ)-BK(4,JJ)*P2)

-DH(5)*(FK(5,JJ)*Y(ICOL*4+4+JJ)*Y(ICOL*2+2+JJ)-BK(5,JJ)*APR)
                            CONTINUE
13
                            ĎŐ 15 L=2, IČŐL
F2(L)=(TK/R**2.6)*((TP(L+1)-TP(L-1))/2.6/DR
1 /COL(L)+(TP(L+1)-2.0*TP(L)+TP(L-1))/(DR**2.0))
                            T /CUL(L)+(TP(L+1)-2.0*TP(L)+TP(L-1))/(DR**2.0
CONTINUE
TYPE *,F1(ICOL+1),F2(ICOL+1)
DO 17 J=1,ICOL-1
F(ICOL*6+6+J)=(F2(J+1)+RHO(J+1)*F1(J+1)/1000.0)
2 /(RX(J+1)*RHO(J+1)*CP(J+1))
IF(ICOND.EQ.0) F(ICOL*6+6+J)=0.0
CONTINUE
DETURN
15
C
 17
                            RETURN
                            END
```

```
************************
                                                 SUBROUTINE OUTPUT
 45
55
17
16
                                             U6(J)=U(J)*RY(6,J,IT)

U7(J)=U(J)*(RY(7,J,IT)+273.)*CP(J)

CONTINUE

IPT=ICOL+1

IFAIL1=0

CALL D01GAF(COL,U,IPT,SUMM,ER,IFAIL1)

CALL D01GAF(COL,U1,IPT,SUMM,ERR,IFAIL1)

CALL D01GAF(COL,U2,IPT,SUMM,ERR,IFAIL1)

CALL D01GAF(COL,U3,IPT,SUMM,ERR,IFAIL1)

CALL D01GAF(COL,U3,IPT,SUMM,ERR,IFAIL1)

CALL D01GAF(COL,U4,IPT,SUMM,ERR,IFAIL1)

CALL D01GAF(COL,U4,IPT,SUMM,ERR,IFAIL1)

CALL D01GAF(COL,U4,IPT,SUMM,ERR,IFAIL1)

CALL D01GAF(COL,U6,IPT,SUMM,ERR,IFAIL1)

CALL D01GAF(COL,U6,IPT,SUMM,ERR,IFAIL1)

IF(IFAIL,NE.0) G0 T0 99

U1AV=SUMM/SUMM

TY(10,IT)=SUMM/SUMM

TY(11,IT)=SUMM/SUMM

TY(11,IT)=SUMM/SUMM

TY(13,IT)=SUMM/SUMM

TY(13,IT)=SUMM/SUMM

TY(13,IT)=SUMM/SUMM

TY(13,IT)=SUMM/SUMM

TY(13,IT)=SUMM/SUMM

TY(13,IT)=SUMM/SUMM-273.

TY(14,IT)=SUMM/SUMM-273.

TY(14,IT)=SUMM/SUMM-273.

TY(14,IT)=SUMM/SUMM-273.

WRITE(39,23) (RY(2,I,IT),I=1,ICOL+1)

WRITE(39,23) (RY(2,I,IT),I=1,ICOL+1)

WRITE(39,23) (RY(2,I,IT),I=1,ICOL+1),(RY(3,I,IT),I=1,ICOL+1)

1 (RY(4,I,IT),I=1,ICOL+1),(RY(6,I,IT),I=1,ICOL+1)

1 (RY(4,I,IT),I=1,ICOL+1),(RY(6,I,IT),I=1,ICOL+1)

FORMAT(10X,P(1)::,20F10.6,/,5X,Oth MOMENT::,20F10.6,/,

1 5X, ist MOMENT::,20F10.6,/,9X,DIMER::,20F10.6,/,
18
 22
 39
```

TX(IT)=TOUT IFAIL=1

```
C
65
90
13
      C
      63
C
C
      695
64
      RY(VARIABLE NO., MESH PT NO., IT)
TY(VARIABLE NO., IT)
TY(I,J): 1: 10=AV CONV; 11=AV CHAIN LENGTH
12=AV WATER; 13=AV TEMP(C)
14=AV DIMER
      DŮ 28 J=1, NPTS
RY(7,J,IT)=TEMP(7,J)
```

```
28
22
39
23
24
26
29
47
27
C
691
101
103
102
104
105
107
106
108
60
19
21
693
109
CC
CC
692
        "ÔŬŦPŪŦ" FOR "PLŌŤŤĬŇĞ"
       WRITE(55,*) (X(1),1=1,NPTS),(TX(J),J=1,L)
```

```
WRITE(55,*) (((RY(I,J,K),I=1,9),J=1,NPTS),K=1,L)
WRITE(55,*) ((TY(I,J),I=10,14),J=1,L)
           STOP
TYPE *, IFAIL
TYPE *, IFAIL, TOUT
STOP
99
100
           CCC
           C
          IF(X.EQ.B) C(I)=1.E-04
G(I)=0.5E-20
34
35
           DS(5)=5.8265E-01

AC(1)=4.3075E+07

AC(2)=1.2114E+10

AC(3)=1.6377E+10

AC(4)=2.3307E+12

AC(5)=3.011E+09

EC(1)=1.8806E+04

EC(2)=2.067E+04

EC(3)=2.0107E+04

EC(4)=3.74E+04

UR=1.987

DU 11 I=1.5

FK(1)=A0(1)*EXP(-E0(1)/(UR*UX(7)))+AC(1)*EXP(-EC(1)/(UP*10X(7)))*UX(3)

EK(1)=EXP(DS(1)/UR-DH(1)/(UR*UX(7)))

BK(1)=FK(1)/EK(1)

CONTINUE
C
           CONTINUE
11
```

```
P2=UX(2)
P3=UX(2)
APR=(UX(3)-2.*UX(2))
IF(APR.LT.0.0) APR=0.0
IF(APR.LT.0.0) P2=0.0
IF(APR.LT.0.0) P3=0.0
 C
                 F(1)=-FK(1)*0x(1)*0x(6)+BK(1)*0x(2)-FK(3)*0x(1)*0x(3)
4 +BK(3)*(Ux(3)-Ux(2))
 C
                F(2)=FK(1)*UX(1)*UX(6)-BK(1)*UX(2)-2.*FK(2)*UX(2)*UX(3)

1 +2.*BK(2)*UX(6)*(UX(3)-UX(2))-FK(3)*UX(2)*UX(1)+

BK(3)*P2-FK(5)*UX(2)*UX(5)+BK(5)*P3
 C
                F(3)=FK(1)*UX(1)*UX(6)-BK(1)*UX(2)-FK(2)*UX(3)*UX(3)*
BK(2)*UX(6)*(UX(4)-UX(3))+FK(4)*UX(5)*UX(6)-
BK(4)*P2
 C
                F(4)=FK(1)*UX(1)*UX(6)-BK(1)*UX(2)+FK(3)*UX(1)*UX(3)-

BK(3)*(UX(3)-UX(2))+2.*FK(5)*UX(5)*UX(3)-2.*BK(5)*

APR+2.*FK(4)*UX(6)*UX(5)-2.*BK(4)*P2
 C
                P(5)=-FR(4)*UX(5)*UX(6)+BR(4)*P2-FR(5)*UX(5)*UX(3)+
BK(5)*APR
C
                  (6)=-FK(1)*UX(1)*UX(6)+BK(1)*UX(2)+FK(2)*UX(3)*UX(3)*UX(3)-BK(2)*UX(6)*(UX(4)-UX(3))-FK(4)*UX(5)*UX(6)+BK(4)*P2
                12
C
                Fi=-DH(i)*(FK(i)*UX(i)*UX(6)-BK(i)*UX(2))

-DH(2)*(FK(2)*UX(3)*UX(3)-BK(2)*UX(6)*(UX(4)-UX(3)))

-DH(3)*(FK(3)*UX(1)*UX(3)-BK(3)*UX(3)+BK(3)*UX(2))

-DH(4)*(FK(4)*UX(5)*UX(6)-BK(4)*P2)

-DH(5)*(FK(5)*UX(5)*UX(3)-BK(5)*APR)
               DO 12 I=1,7
IF(UX(1).LT.0.0) UX(1)=8.8
CONTINUE
IF(ICOND.E0.0) GO TO 55
F(7)=F1*RHO/1000.6
GO TO 66
F(7)=0.
RETURN
C
C
C
C12
55
66
                CCC
                SUBROUTINE BNDY *********************************
               C
 44
C
                RIGHT HAND B.C.
                CONTINUE
DO 55 I=1,6
P(I)=0.
O(I)=1.
E(I)=0.
 20
                CONTINUE
P(7)=0
Q(7)=-TK
 55
```

```
*************
    CCCCC
                                ORTHOGONAL COLLOCATION: METHOD IV
                              ************
   C
   C
    17
                                1 //)
FORMAT(15X,50('*'),///,30X,'ADTABATIC REACTOR',///,15X,50('*'),
                             FORMAT(15X,50('*'),//,30X, ADTABATIC REACTOR WITH HEAT

1 //
FORMAT(15X,50('*'),//,22X, TUBULAR PEACTOR WITH HEAT

1 TRANSFER',//,15X,50('*'),/)
WRITE(39,20) R,ICOL,U10,U40,U20,U50,U70,U30,U60
FORMAT(5X, RADIUS=',F6.2,5X,'80 OF COLL PUINTS=',I3,/,

5X,'MONOMER=',F8.4,5X,'1st WOMENT=',F8.4,5X,/,

2 8X,'P(1)=',F8.4,10X,'DIMER=',F8.4,10X,'TEMP=',F10.4,/,

312X,'Oth MOMENT=',F8.4,10X,'WATER=',F8.4,/,75('*'))
TH=TM-273.
IF(ICOND,E0.2) WRITE(39,21) U0,TM
FORMAT(7,5X,'HEAT TRANSF COEFF=',F10.5,2X,'KCAL/(SO M)(HR)

1 (DEG K)',/,10X,'COOLANT TEMP=',F10.5,//,75('*'))
TOL=1.E-06
   18
   19
20
   21
                              TOL=1.E-06
IRELAB=2
MPED=0
                              MPED=0
X=0.0
H=0.5
IT=0
D0 33 J=1,ICOL+1
RX(J)=2.*(1.-(COL(J))**2.)
IF(RX(J).E0.0.0) RX(J)=1.E-04
IF(IVEL.EQ.0) RX(J)=1.0
CONTINUE
INITIAL CONDITION
D0 22 I=1,(ICOL+1)
Y(I)=U10
Y(ICOL+1+I)=U20
Y(ICOL+2+2+I)=U30
   33
                              Y(ICOL*3+3+I)=040

Y(ICOL*4+4+I)=050

Y(ICOL*5+5+I)=060

Y(ICOL*5+5+I)=070/500.0

Y(ICOL*6+6+I)=070/500.0

CONTINUE

IF(IVEL,EQ.0) GO TO 23

Y(ICOL*1)=0.670647

Y(ICOL*2+2)=0.000234

Y(ICOL*3+3)=0.043585

Y(ICOL*4+4)=8.65944

Y(ICOL*5+5)=0.034956

Y(ICOL*5+5)=0.116415

IFAIL=0

N=(ICOL+1)*7-1
   22
   23
```

```
IW=18+N
TW=TW/500.0
CALL D02EBF(X,XEND,N,Y,TOL,IRELAB,FCN,MPED,
PEDERV,OUTPUT,W,IW,IFAIL)
WRITE(39,26) (COL(I),I=1,ICOL+1)
FORMAT(10X,10F8.3)
IF(IFAIL.NE.0) G0 TO 24
STOP
26
                      STOP
TYPE *, IFAIL
24
                    END
                      *******************
CCC
10
44
16
                     DS(5)=5.8265E-01

AC(1)=4.3075E+07
AC(2)=1.2114E+10
AC(3)=1.6377E+10
AC(4)=2.3307E+12
AC(5)=3.011E+09
EC(1)=1.8806E+04
EC(2)=2.067E+04
EC(3)=2.067E+04
EC(4)=3.74E+04
EC(5)=2.04E+04
UR=1.987
DO 11 J=1,ICOL+1
FK(I,J)=AO(I)*EXP(-EO(I)/(UR*TP(J)*500.))+AC(I)*EXP
1 (-EC(I)/(UR*TP(J)*500.0))*Y(ICOL*2+2+J)
EK(I,J)=EXP(DS(I)/UR-DH(I)/(UR*TP(J)*500.0))
BK(I,J)=FK(I,J)/EK(I,J)
CONTINUE
DO 12 J=1,ICOL+1
C
 11
```

```
P2=Y(ICUL+1+J)

P3=Y(ICUL+1+J)

APR=(Y(ICUL*2+2+J)-2.*Y(ICUL+1+J))

IF(APR.LT.0.0) APR=0.0

IF(APR.LT.0.0) P2=0.0

IF(APR.LT.0.0) P3=0.0
C
                          F(J)=-FR(1,J)*Y(J)*Y(ICOL*5+5+J)+BR(1,J)*Y(ICOL+1+J)*
1 -FK(3,J)*Y(J)*Y(ICOL*2+2+J)
2 +BK(3,J)*(Y(ICOL*2+2+J)-Y(ICOL+1+J))
F(J)=F(J)/RX(J)
                       F(icoL+i+j)=FR(i,j)*Y(j)*Y(icoL*5+5+j)-BR(i,j)*Y(icoL+1+J)

1 -2.*FK(2,J)*Y(icoL+1+J)*Y(icoL*2+2+J)

2 +2.*BK(2,J)*Y(icoL*5+5+J)*(Y(icoL*2+2+J)-Y(icoL+1+J))

3 -FK(3,J)*Y(icoL+1+J)*Y(J)+

4 BK(3,J)*P2-FK(5,J)*Y(icoL+1+J)*Y(icoL*4+4+J)+BK(5,J)*P3

F(icoL+i+J)=F(icoL+1+J)/RX(J)
C
C
                         F(ICOL*3+3+J)=FK(I,J)*Y(J)*Y(ICOL*5+5+J)-BK(I,J)*Y(ICOL+1+J)

1 +FK(3,J)*Y(J)*Y(ICOL*2+2+J)-

BK(3,J)*(Y(ICOL*2+2+J)-Y(ICOL+1+J))+2.*FK(5,J)*

Y(ICOL*4+4+J)*Y(ICOL*2+2+J)-2.*BK(5,J)*

APR+2.*FK(4,J)*Y(ICOL*5+5+J)*Y(ICOL*4+4+J)-2.*BK(4,J)*P2

F(ICOL*3+3+J)=F(ICOL*3+3+J)/RX(J)
C
C
                         C
                         F(icoi*5+5+J)=-FK(i;J)*Y(j)*Y(icoi*5+5+J)+BK(i;J)*Y(icob+1+J)

+FK(2,J)*Y(icob*2+2+J)*Y(icob*2+2+J)-

BK(2,J)*Y(icob*5+5+J)*(y(icob*3+3+J)-Y(icob*2+2+J))

-FK(4,J)*Y(icob*4+4+J)*Y(icob*5+5+J)+BK(4,J)
                         2 *P2
F(ICOL*5+5+J)=F(ICOL*5+5+J)/RX(J)
                        CONTINUE
DD 13 JJ=1,ICOL
P2=Y(ICOL+1+JJ)
P3=Y(ICOL+1+JJ)
APR=(Y(ICOL*2+2+JJ)-2.*Y(ICOL+1+JJ))
IF(APR.LT.0.0) APR=0.0
IF(APR.LT.0.0) P3=0.0
IF(APR.LT.0.0) P3=0.0
F1==DH(1)*(FK(1,JJ)*Y(JJ)*Y(ICOL*5+5+JJ)-BK(1,JJ)*Y(ICOL+1+JJ))
1 --DH(2)*(FK(2,JJ)*Y(ICOL*2+2+JJ)*Y(ICOL*2+2+JJ)-BK(2,JJ)*
2 Y(ICOL*5+5+JJ)*(Y(ICOL*3+3+JJ)-Y(ICOL*2+2+JJ)-BK(2,JJ)*
2 Y(ICOL*5+5+JJ)*(Y(ICOL*3+3+JJ)-Y(ICOL*2+2+JJ)-BK(3,JJ)*
4 Y(ICOL*2+2+JJ)+BK(3,JJ)*Y(ICOL*2+2+JJ)-BK(3,JJ)*
5 --DH(4)*(FK(4,JJ)*Y(ICOL*4+4+JJ)*Y(ICOL*5+5+JJ)-BK(4,JJ)*P2)
6 --DH(5)*(FK(5,JJ)*Y(ICOL*4+4+JJ)*Y(ICOL*2+2+JJ)-BK(5,JJ)*APR)
C
12
 C
                          SUM2=0.0

DO 14 L=1,ICOL

SUM2=SUM2+B(JJ,L)*TP(L)

F2=(TK/R**2.)*(SUM2+B(JJ,ICOL+1)*TP(ICOL+1))
 14
                          F(ICOL*6+6+JJ)=(F2+RHO(JJ)*F1/1000.0/500.0)/(RX(JJ)
2 *RHO(JJ)*CP(JJ))
IF(ICOND.EQ.0) F(ICOL*6+6+JJ)=0.0
CONTINUE
RETURN
 13
```

```
SUBROUTINE OUTPUT
55
17
 25
                                                                             IF(ICOND.EQ.0) TLAST=U70/500.0
RY(7,ICOL+1,IT)=TLAST*500.0-273.0

DO 16 J=1,ICOL+1
RHO(J)=1060.0/(1.0065+0.0123*Y(J))+(RY(7,J,IT)+273.0-495.)*
3 (0.00035+0.00007*Y(J)))
(CP(J)=0.6593*Y(J))/10+(1.-Y(J)/U10)*(486.1+0.337*(PY(7,J,IT)+273.0)/10+00.0
CONTINUE
SUM1=0.0;SUM2=0.;SUM3=0.;SUM4=0.;SUM5=0.;SUM6=0.;SUM7=0.
SUM1=0.0;SUMM=0.
DO 18 J=1,ICOL+1
U(J)=RHO(J)*PX(J)
U(J)=RHO(J)*PX(J)
U(J)=RHO(J)*PX(J)
U(J)=U(J)*RY(1,J,IT)
U(J)=U(J)*RY(2,J,IT)
U(J)=U(J)*RY(5,J,IT)
U(J)=U(J)*RY(5,J,IT)
U(J)=U(J)*RY(6,J,IT)
U(J)=U(J)*RY(6,J,IT)
U(J)=U(J)*RY(7,J,IT)+273.)*CP(J)
CONTINUE
SUM4=SUM4+WT(J)*U1(J)
SUMM=SUM4+WT(J)*U1(J)
SUMM=SUM4+WT(J)*U1(J)
SUMM=SUM4+WT(J)*U1(J)
SUMM=SUM4+WT(J)*U1(J)
SUMM=SUM5+WT(J)*U1(J)
SUMM=SUM7+WT(J)*U1(J)
SUMM=SUM7-WT(J)*U1(J)
SUM7-SUM7-WT(J)*U1(J)
SUM7
  16
    18
       19
         22
```

```
39
23
 24
26
29
 47
27
36
35
                     C
                    SUBROUTINE PEDERV(T,Y,PW)
DIMENSION Y(70),PW(70,70)
RETURN
END
SUBROUTINE COLLOC(N,X,A,B,W)
IMPLICIT REAL *8 (A-H,O-Z)
dimension qinv(i0,10),z(10),c(10,10),d(10,10)
dimension a(20,20),b(20,20),g(20,20),x(20),w(20)
DIMENSION DIF1(10),DIF2(10),DIF3(10),ROOT(10)
COMMON/ONE/ X1(20)
N=2
NT=N+1
BETA=0.0
BETA=0.0
BETA=0.0
33 K=1,N
K=1,N
CONT(NT)=1.0
DO 33 K=1,N
X1(K)=(SQRT(ROOT(K)))
TYPE *,X1(K)
CONTINUE
Call coll(a,b,q,x,w,20,N,aa)
TYPE *,(X(I),I=1,NT)
TYPE *,((A(I,J),J=1,NT),I=1,NT)
TYPE *,((A(I,J),J=1,NT),I=1,NT)
TYPE *,((B(I,J),J=1,NT),I=1,NT)
TYPE *,((B(I,J),J=1,NT),I=1,NT)
TYPE *,(W(I),I=1,NT)
RETURN
STOP
C
Ç
C
33
CCCC
C
                      STOP
                     ENO
SUBROUTINE JCOBI (ND,N,NO,N1,AL,BE,DIF1,DIF2,DIF3,ROOT)
IMPLICIT REAL *8 (A-H,O-Z)
C
                      DIMENSION DIF1(10), DIF2(10) , DIF3(10), ROOT(10)
0000000
                        EVALUATION OF ROOTS AND DERIVATIVES OF JAACUBI POLYNOMIALS P(N) (AL*BE); MACHINE ACCURACY 16 D;
                        FIRST EVALUATION OF COEFFICIENT IN RECURSION FORMULAS RECURSION COEFFICIENTS ARE STORED IN DIF1 AND DIF?
                     AB=AL+BE

AD=BE-AL

AP=BE*AL

DIF1(1)=((AD/(AB+2.))+1.)/2.

DIF2(1)=0.

IF (N.LT. 2) GO TO 15

DO 10 1=2,N

71=1-1
                      Z1=1-1
Z=AB+2*Z1
```

```
DIF1(I)=(((AB*AD)/(Z*(Z+2.)))+1.)/2.
IF (I.NE.2) GO TO 11
DIF2(I)=((AB+AP+Z1)/((Z+1)*Z*Z))
GO TO 10
Z=Z*Z
Y=Z1*(AB+Z1)
Y=Y*(AP+Y)
DIF2(I)=Y/((Z-1)*Z)
CONTINUE
  11
  10
CC
CC
15
                                ROOT DETERMINAATION BY NEWTON METHOD WITH SUPPRESSION OF PREVIOUS
                           X=0
DO 20
XD=0.
XN=1.
                                                  I=1, N
  25
                          XD1=0.

XN1=0.

DO 30 J=1.N

XP=(-DIF1(J)+X)*XN-DIF2(J)*XD

XP1=(-DIF1(J)+X)*XN1-DIF2(J)*XD1+XN

XD=XN

XD1=XN1

XN=XP

XN1=XP1

CONTINUE

ZC=1.

Z=XN/XN1

IF (I EQ 1) GO TO 21

DO 22 J=2,I
  30
                           ZC=ZC-Z/(X-ROOT(J-1))
Z=Z/ZC
X=X-Z
IF (DABS(Z).GT.1.D-09).GO TO 25
IF(ABS(Z).GT.1.E-06).GO TO 25
ROOT(I)=X
X=X+.0001
CONTINUE
DIF3(I)=0.
RETURN
END
SUBROUTINE COLL(A.B.O.X.W.ND.N.
  22
21
· C
  20
                            END
SUBROUTINE COLL(A,B,O,X,W,ND,N,AA)
IMPLICIT REAL *8 (A-H,O-Z)
DIMENSION A(ND,ND),B(ND,ND),O(ND,ND),X(ND),W(ND)
DIMENSION OINV(10,10),Z(10),C(10,10),D(10,10)
COMMON/ONE/X1(20)
  C
                            N1=N+1
  CCC
                            THIS SUBROUTINE COMPUTES THE USING SYMMETRIC POLYNOMIALS DO 57 I=1,N X(I)=X1(I) CONTINUE X(N1)=1.0 DO 30 I=1,N1
                                                                                                                                MATRICES FOR ORTHOGONAL COLLOCATION
   57
                           DU 30 1-2,000
AI=I
Z(I)=1./(2.*AI+AA-2.)
CONTINUE
DO 35 I=1,N1
O(I,1)=1.
OINV(I,1)=1.
DO 35 J=2,N1
   30
                            O(I,J)=X(I)**(2*J-2)
OINV(I,J)=O(I,J)
DO 40 J=1,N1
CA=2.*J-2.
   35
```

```
DA=(2.*J-2.)*(2.*J+AA-4.)

DO 40 I=1,N1

C(I,J)=CA*X(I)**(2*J-3)

D(I,J)=DA*X(I)**(2*J-4)

CONTINUE

CALL INVR (OINV,N1,10)

DO 50 I=1,N1

A(I,J)=0.0

DO 45 K=1,N1

A(I,J)=A(I,J)+C(I,K)*OINV(K,J)

B(I,J)=B(I,J)+D(I,K)*QINV(K,J)

CONTINUE

O(I,J)=OINV(I,J)

W(I)=W(I)+Z(J)*QINV(J,I)

CONTINUE

WRITE(60,12) (X(I),I=1,N1)

FORMAT(3E16.5)

FORMAT(BIJ VALUES ARE ')
  40
  45
 50
C
 132
                                            WRITE(60,12)
WRITE(61,55)((A(I,J),I=1,N1),J=1,N1)
WRITE(60,13)
WRITE(62,55)((B(I,J),I=1,N1),J=1,N1)
FORMAT (9E16.5/)
RETURN
FORM
00005
                                              END
                                          SUBROUTINE INVR(A,N,NI)
IMPLICIT REAL *8 (A-H,O-Z)
DIMENSION A(NI,NI),B(20),C(400)
DO 5 J=1,N
IND=N*(J-1)
DO 5 I=1,N
C(IND+I)=A(I,J)
CONTINUE
CALL INVERT(N,1,C,B,1)
DO 20 J=1,N
B(I)=0.
B(J)=1.
CALL INVSW(N,1,C,B,1)
DO 15 I=1,N
A(I,J)=B(I)
CONTINUE
RETURN
END
CHEROUTINE INVERT(N,NE A, B, ITVE
C
5
 10
15
                                            END
SUBROUTINE INVERT(N, NE, A, B, ITYPE)
IMPLICIT REAL *8 (A-H, O-Z)
C
                                            DIMENSION A(1000,3),B(N),A1(1000),B1(1000),C1(1000)
GO TO (5,10,15),ITYPE
CALL DECOMP(N,A)
RETURN
NP=(N-1)/NE+1
PAUSE 14
CALL LUDECO(NP,NE,A)
RETURN
DO 20 K=1,N
A1(K)=A(K,1)
B1(K)=A(K,2)
C1(K)=A(K,3)
CONTINUE
CALL INVTRI(N,A1,B1,C1)
DO 25 K=1,N
A(K,1)=A1(K)
A(K,2)=B1(K)
A(K,2)=B1(K)
A(K,3)=C1(K)
 5
 10
 15
 20
```

```
CONTINUE
RETURN
ENTRY INVSW
GO TO (30,35,40),ITYPE
CALL SOLVE(N,A,B)
RETURN
25
30
                         CALL FAS(NP, NE, N, A, B)
RETURN
CALL SWEEP(N, A1, B1, C1, B)
RETURN
35
40
                         END
Ç
                         SUBROUTINE SING(I)
IMPLICIT REAL *8 (A-H,0-Z)
GO TO (5,15),I
WRITE(60,10)
FORMAT ('MATRIX WITH ZERO ROW DECOMPOSE')
C
5
10
                         RETURN
WRITE(60,20)
FORMAT( SINGULAR MATRIX IN DECOMPOSE. ZERO IN SOLVE
15
20
                         RETURN
END
SUBROUTINE DECOMP(N,A).
IMPLICIT REAL *8 (A-H,O-Z)
DIMENSION A(N,N)
COMMON /DENSE/ IPS(201),SC(201)
C
Ç
                        IF (N .EQ. 1) RETURN
DO 25 I=1,N
IPS(I)=I
ROWNRM=0.0
DO 10 J=1,N
IF (ROWNRM-ABS(A(I,J))) 5,10,10
ROWNRM = ABS(A(I,J))
CONTINUE
5
                         CONTINUE
IF (ROWNRM) 15,20,15
SC(I)=1,/ROWNRM
GO TO 25
15
C
20
                         CALL SING(1)
SC(1)=0.
CONTINUE
25
C
                         UM1=N-1
DO 65 K=1,NM1
                         DO 65 K=1,NM1
BIG=0.

DO 35 I=K,N
IP=IPS(I)
SIZE=ABS(A(IP,K)*SC(IP))
IF (SIZE-BIG) 35,35,30
BIG=SIZE
IDXPIV=I
CONTINUE
IF (BIG) 45,40,45
30
35
                         IF (BIG) 45,40,45
CALL SING(2)
GO TO 65
40
                         IF (IDXPIV-K) 50,55,50

J=IPS(K)
IPS(K)=IPS(IDXPIV)
IPS(IDXPIV)=J
KP=IPS(K)
PIVOT=A(KP,K)
KP1=K+1
DO 60 I=KP1,N
IP=IPS(I)
EM=-A(IP,K)/PIVOT
A(IP,K)=-EM
DO 60 J=KP1,N
45<sub>0</sub>
55
```

```
A(IP,J)=A(IP,J)+EM*A(KP,J)
CONTINUE
CONTINUE
KP=IPS(N)
IF (A(KP,N)) 75,70,75
CALL SING(2)
CONTINUE
RETURN
END
60
65
70
75
CC
                               SUBROUTINE SOLVE(N,A,B)
IMPLICIT REAL *8 (A-H,O-Z)
DIMENSION A(N,N),B(N)
COMMON /DENSE/ IPS(201),SC(201)
C
C
                               IF (N .GT. 1) GO TO 5
B(1)=B(1)/A(1,1)
BETURN
                             RETURN
CONTINUE
NP1=N+1
IP=IPS(1)
SC(1)=B(IP)
DD 15 I=2,N
IP=IPS(1)
IM1=I-1
SUM=SUM+A(IP,J)*SC(J)
CONTINUE
SC(I)=B(IP)-SUM
CONTINUE
IP=IPS(N)
SC(N)/A(IP,N)
DC(N)-IBACK=2,N
I=NP1-IBACK
IP=IPS(I)
IP1=I+1
SUM=20.
I-TD1
5
10
 15
                               SUM=0.
DD 20 J=IP1.N
SUM=SUM+A(IP,J)*SC(J)
                               SUM=SUM+A(IP,J)*SC(J)
CONTINUE
SC(I)=(SC(I)-SUM)/A(IP,I)
CONTINUE
DO 30 I=1,N
B(I)=SC(I)
RETURN
END
 20
 25
 30
 C
                                SUBROUTINE FAS(NP, NE, NT, A, B)
IMPLICIT REAL *8 (A-H, O-Z)
DIMENSION A(NP, NP, NE), B(NT)
 C
 C
                               NP1=NP
DO 10 L=1,NE
DO 10 I=2,NP1
I2=(L-1)*(NP-1)+I
S=0.
I1=I-1
DO 5 J=1,I1
J2=I2-I+J
S=S+A(I,J,L)*B(J2)
CONTINUE
B(I2)=B(I2)-S
CONTINUE
  5
  10
                                DO 25 L1=1,NE
L=NE-L1+1
IF (L.NE.NE) GO TO 15
```

```
B(NT)=B(NT)/A(NP,NP,NE)
N1=NP-1
DO 25 K=1,N1
I=N1+1-K
I2=(L-1)*(NP-1)+I
15
                         I2=(L-1)*(NP-1)+I

M=I+1

N2=N1+1

S=0.

DO 20 J=M,N2

J2=(L-1)*(NP-1)+J

S=S+A(I,J,L)*B(J2)

B(I2)=(B(I2)-S)/A(I,I,L)

CONTINUE

RETURN
20
 25
                          END
C
                         SUBROUTINE LUDECO(NP,NE,A)
IMPLICIT REAL *8 (A-H,O-Z)
DIMENSION A(NP,NP,NE)
C
                         H1=HP-1
D0 10 L=1,NE
D0 5 K=1,H1
F1=K+1
D0 5 I=K1,NP
S=A(I,K,L)/A(K,K,L)
A(I,K,L)=S
D0 5 J=K1,NP
A(I,J,L)=A(I,J,L)-S*A(K,J,L)
CONTINUE
IF (L.EO.NE) RETURN
A(1,L+1)=A(NP,NP,L)
CONTINUE
END
C
 5
 10
 C
                          SUBROUTINE INVTRI(N,A,B,C)
IMPLICIT REAL *8 (A-H,O-Z)
DIMENSION A(N),B(N),C(N)
 C
 C
                          DO 5 L=2,N
S=A(L)/B(L-1)
B(L)=B(L)-S*C(L-1)
A(L)=S
RETURN
 5
                           END
 C
                           SUBROUTINE SWEEP(N,A,B,C,D)
IMPLICIT REAL *8 (A-H,0-Z)
DIMENSION A(N),B(N),C(N),D(N)
 C
 C
                           DO 5 L=2, N D(L)=D(L)-A(L)*D(L-1)
 5
C
                           D(N)=D(N)/B(N)
DD 10 L=2,N
K=N-L+1
D(K)=(D(K)-C(K)*D(K+1))/B(K)
RETURN
  10
                           END
 CC
```